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RESEARCH STUDY ON STABILIZATION AND CONTROL MODERN SAMPLED-DATA CONTROL THEORY

SYSTEMS RESEARCH LABORATORY

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PREPARED FOR GEORGE C. MARSHALL SPACE FLIGHT CENT



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subtitle:

CONTINUOUS AND DISCRETE

DESCRIBING FUNCTION ANALYSIS

OF THE IPS SYSTEM

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I. MODEL DEVELOPMENT OF THE CONTINUOUS-DATA IPS CONTROL SYSTEMS

The objective of this chapter is to develop the dynamic equations and the mathematical model of the continuous-data IPS control system. The IPS model considered includes one flexible body mode and is hardmounted to the Orbiter/Pallet. The model contains equations describing a torque feed-forward loop (using accelerometers as inputs) which will aid in reducing the pointing errors caused by Orbiter disturbances.

The equations of motion of the IPS system are written as

$$MR + DR + KR + Q Rdt = F$$
 (1-1)

where

$$R = \begin{bmatrix} X_s \\ Z_s \\ \theta_s \\ \theta_i \\ X \\ Z \\ \eta \end{bmatrix}$$
 (1-2)

 X_s , Z_s : translations of the Orbiter

 θ_s : attitude of the Orbiter

 θ : attitude of the instrument

X, Z: accelerometer measurements

M: 7 × 7 mass matrix

D: 7×7 damping matrix

K: 7 × 7 stiffness matrix

Q: 7 × 7 integral control matrix

F: 7 × 1 generalized force vector

The elements of M are:

$$\begin{array}{l} m_{11} = m_{0} + m_{1} = 91,200 \\ m_{13} = m_{1} (d_{sm} + c)_{z} = -226 \\ m_{14} = m_{1} (m_{bz}c + c)_{x} = 16,600 \\ m_{22} = m_{0} + m_{1} = 91,200 \\ m_{23} = -m_{1} (d_{sm} + c)_{x} = 16,600 \\ m_{24} = -m_{1} (-r_{bz}S + r_{bx}C + c)_{z} = -226 \\ m_{31} = m_{13} = m_{1} (d_{sm} + c)_{z} = -226 \\ m_{32} = m_{23} = 16,600 \\ m_{33} = 1_{0y} + 1_{1y} + m_{1} ((d_{sm} + c)_{x}^{2} + (d_{sm} + c)_{z}^{2}) = 7.06 \times 10^{6} \\ m_{34} = 1_{1y} + m_{1} ((d_{sm} + c)_{z} (r_{bz}C + r_{bx}S + c)_{x} + (d_{sm} + c)_{x} (-r_{bz}S + r_{bx}C + c)_{z} \\ = 26,700 \\ m_{41} = m_{14} = 2,270 \\ m_{42} = m_{24} = 3,920 \\ m_{43} = m_{34} = 26,700 \\ m_{44} = 1_{1y} + m_{1} (r_{bx}^{2} + r_{bz}^{2}) = 10,300 \\ m_{53} = -d_{smz} = 0.929 \\ m_{55} = 1.0 \\ m_{57} = h_{x} = -0.00113 \\ m_{63} = d_{smx} = -4.72 \\ m_{66} = 1.0 \\ m_{67} = h_{z} = 0.00137 \\ m_{77} = 1.0 \\ \end{array}$$

All other elements of M are zero.

 $m_{62} = -1$

The elements of D are:

$$d_{43} = d_{44} = K_r = 19,700$$

$$d_{47} = K_r \sigma_{rg} = -76.4$$

$$d_{55} = 2\zeta_x \omega_x = 44$$

$$d_{66} = 2\zeta_z \omega_z = 44$$

 $d_{77} = 2\zeta_b \omega_b = 0.115$

All other elements of D are zero.

The elements of K are:

$$k_{43} = k_{44} = K_p = 70,000$$
 $k_{45} = -m_i (r_{bz} C\phi + r_{bx} S\phi) \omega_{ay}^2 = -2.23 \times 10^6$
 $k_{46} = m_i (-r_{bz} S\phi + r_{bx} C\phi) \omega_{ay}^2 = -3.87 \times 10^6$
 $k_{47} = K_p \sigma_{ss} = -272$
 $k_{55} = \omega_x^2 = 986$
 $k_{66} = \omega_z^2 = 986$
 $k_{77} = \omega_b^2 = 132$

All other elements of K are zero.

The elements of Q are:

$$q_{43} = q_{44} = K_i = 1.1 \times 10^5$$

 $q_{47} = K_i \sigma_{ss} = -427$

All other elements of Q are zero.

The generalized force vector is

$$F = [F_{x} F_{z} r_{cx}F_{z} - r_{cz}F_{x} 0 0 0 h_{zc}F_{z} + h_{xc}F_{x}]'$$
 (1-3)

Given the elements of the matrices M, D, K and Q, the equations of motion of Eq. (1-1) are written as

$$m_{11}\ddot{X}_{S} + m_{13}\ddot{\Theta}_{S} + m_{14}\ddot{\Theta}_{I} = F_{X}$$
 (1-4)

$$m_{22}\ddot{z}_s + m_{23}\ddot{\Theta}_s + m_{24}\ddot{\Theta}_i = F_z$$
 (1-5)

$$m_{31}\ddot{X}_{s} + m_{32}\ddot{Z}_{s} + m_{33}\ddot{\Theta}_{s} + m_{34}\ddot{\Theta}_{i} = r_{cx}F_{z} - r_{cz}F_{x}$$
 (1-6)

$$m_{41}\ddot{X}_{s} + m_{42}\ddot{Z}_{s} + m_{43}\ddot{\Theta}_{s} + m_{44}\ddot{\Theta}_{i} + K_{r}\dot{\Theta}_{s} + K_{r}\dot{\Theta}_{i} + K_{r}\sigma_{rg}\dot{\eta}$$

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$$+ \kappa_{p} \Theta_{s} + \kappa_{p} \Theta_{i} + k_{45} X + k_{46} Z + k_{47} \eta + \kappa_{i} \int_{\Theta_{s}} dt$$

+
$$K_i \int \Theta_i dt + K_i \sigma_{ss} \int \eta dt = 0$$
 (1-7)

$$m_{51}\ddot{x}_s + m_{53}\ddot{\Theta}_s + m_{55}\ddot{x} + m_{57}\ddot{\eta} + d_{55}\dot{x} + k_{55}x = 0$$
 (1-8)

$${}^{m}_{62}\ddot{z}_{s} + {}^{m}_{63}\ddot{o}_{s} + {}^{m}_{66}\ddot{z} + {}^{m}_{67}\ddot{\eta} + {}^{d}_{66}\dot{z} + {}^{k}_{66}z = 0$$
 (1-9)

$$m_{77}^{\ddot{\eta}} + d_{77}^{\dot{\eta}} + k_{77}^{\eta} = h_{zc}F_z + h_{xc}F_x$$
 (1-10)

Without the nonlinear wire-cable and flex-pivot torques, the control torque is expressed as

$$T_{c} = -K_{r} \dot{\Theta}_{s} - K_{r} \dot{\Theta}_{i} - K_{p} \Theta_{s} - K_{p} \Theta_{i} - K_{i} \int \Theta_{s} dt - K_{i} \int \Theta_{i} dt$$

$$- k_{45} X - k_{46} Z \qquad (1-11)$$

where K_r , K_p , and K_i are the rate, position, and integral constants of the controller, respectively.

The nonlinear torque due to the flex pivot and wire cable can be lumped into one operator N which operates on Θ_i . Thus, the component, $-N(\Theta_i)$, should be added to Eq. (1-11); i.e.,

$$T_{c} = -K_{r} \dot{\Theta}_{s} - K_{r} \dot{\Theta}_{i} - K_{p} \Theta_{s} - K_{p} \Theta_{i} - K_{i} \int \Theta_{s} dt - K_{i} \int \Theta_{i} dt$$

$$- k_{45} X - k_{46} Z - N(\Theta_{i})$$
(1-12)

Solving for X_s from Eq. (1-4), we have

$$\ddot{X}_{S} = -\frac{m_{13}}{m_{11}} \ddot{\Theta}_{S} - \frac{m_{14}}{m_{11}} \ddot{\Theta}_{i} + \frac{1}{m_{11}} F_{X}$$
 (1-13)

Solving for \ddot{Z}_s from Eq. (1-5), we have

5

$$\ddot{z}_{s} = -\frac{m_{23}}{m_{22}} \ddot{\Theta}_{s} - \frac{m_{24}}{m_{22}} \ddot{\Theta}_{i} + F_{z} \frac{1}{m_{22}}$$
 (1-14)

Substitute Eq. (1-13) and (1-14) into Eq. (1-6), we have

$$-\frac{m_{31}^{m}_{11}}{m_{11}}\ddot{\Theta}_{s} - \frac{m_{31}^{m}_{14}}{m_{11}}\ddot{\Theta}_{i} + \frac{m_{31}}{m_{11}}F_{x} - \frac{m_{32}^{m}_{23}}{m_{22}}\ddot{\Theta}_{s} - \frac{m_{32}^{m}_{24}}{m_{22}}\ddot{\Theta}_{i} + \frac{m_{32}}{m_{22}}F_{z}$$

$$+ m_{33}\ddot{\Theta}_{s} + m_{34}\ddot{\Theta}_{i} = r_{cx}F_{z} - r_{cz}F_{x}$$
(1-15)

The last equation is rearranged to the following form:

$$\left(m_{33} - \frac{m_{31}^{m} 13}{m_{11}} - \frac{m_{32}^{m} 23}{m_{22}}\right) \ddot{\Theta}_{s} + \left(m_{34} - \frac{m_{31}^{m} 14}{m_{11}} - \frac{m_{32}^{m} 24}{m_{22}}\right) \ddot{\Theta}_{i}$$

$$= r_{cx} F_{z} - r_{cz} F_{x} - \frac{m_{31}}{m_{11}} F_{x} - \frac{m_{32}}{m_{22}} F_{z} \tag{1-16}$$

Substitute Eqs. (1-13) and (1-14) into Eq. (1-7), and making use of Eq. (1-12), we have

$$m_{41} \left(\frac{-m_{13}}{m_{11}} \ddot{\Theta}_{s} - \frac{m_{14}}{m_{11}} \ddot{\Theta}_{i} + \frac{1}{m_{11}} F_{x} \right) + m_{42} \left(\frac{-m_{23}}{m_{22}} \ddot{\Theta}_{s} - \frac{m_{24}}{m_{22}} \ddot{\Theta}_{i} + F_{z} \frac{1}{m_{22}} \right) + m_{43} \ddot{\Theta}_{s} + m_{44} \ddot{\Theta}_{i} + K_{r} \sigma_{rq} \dot{\eta} + k_{47} \eta + K_{i} \sigma_{ss} \int \eta dt - T_{c} = 0$$
(1-17)

Simplifying, the last equation becomes

$$\left(m_{43} - \frac{m_{13}^{m} + 1}{m_{11}} - \frac{m_{42}^{m} + 23}{m_{22}}\right) \ddot{\Theta}_{s} + \left(m_{44} - \frac{m_{14}^{m} + 1}{m_{11}} - \frac{m_{42}^{m} + 24}{m_{22}}\right) \ddot{\Theta}_{i} + K_{r}\sigma_{rg}\dot{\eta} + k_{47}\eta + K_{i}\sigma_{ss} \int \eta dt - T_{c} = -\frac{m_{41}}{m_{11}} F_{x} - \frac{m_{42}}{m_{22}} F_{z} \tag{1-18}$$

Similarly, Eq. (1-9) is written as

$${}^{m}_{62}\left(\frac{{}^{m}_{23}}{{}^{m}_{22}}\ddot{\circ}_{s} - \frac{{}^{m}_{24}}{{}^{m}_{22}}\ddot{\circ}_{i} + \frac{1}{{}^{m}_{22}}F_{z}\right) + {}^{m}_{63}\ddot{\circ}_{s} + {}^{m}_{66}\ddot{z} + {}^{m}_{67}\ddot{\eta} + {}^{d}_{66}\dot{z} + k_{66}z = 0$$
(1-19)

Equations (1-16), (1-18), (1-19), (1-8) and (1-10) are now written as

$$M_{s}\ddot{\Theta}_{s} + M_{i}\ddot{\Theta}_{i} = -\left(r_{cz} + \frac{m_{31}}{m_{11}}\right)F_{x} + \left(r_{cx} - \frac{m_{32}}{m_{22}}\right)F_{z}$$
 (1-20)

$$M_{k}\ddot{\Theta}_{s} + M_{n}\ddot{\Theta}_{i} + K_{r}\sigma_{rg}\dot{\eta} + k_{47}\eta + K_{i}\sigma_{ss}\int \eta dt - T_{c} = -\frac{m_{41}}{m_{11}}F_{x} - \frac{m_{42}}{m_{22}}F_{z}$$
 (1-21)

where

$$M_{s} = M_{33} - \frac{m_{13}^{m_{31}}}{m_{11}} - \frac{m_{32}^{m_{23}}}{m_{22}}$$
 (1-22)

$$M_{i} = M_{34} - \frac{M_{31}^{m_{14}} - \frac{M_{32}^{m_{24}}}{M_{11}} - \frac{M_{32}^{m_{24}}}{M_{22}}$$
 (1-23)

$$M_{k} = M_{43} - \frac{m_{13}^{m} 41}{m_{11}} - \frac{m_{42}^{m} 23}{m_{22}}$$
 (1-24)

$$M_{n} = M_{44} - \frac{m_{14}^{m_{41}}}{m_{11}} - \frac{m_{42}^{m_{24}}}{m_{22}}$$
 (1-25)

$$M_p\ddot{\Theta}_s + M_r\ddot{\Theta}_i + m_{66}\ddot{Z} + m_{67}\ddot{\eta} + d_{66}\dot{Z} + k_{66}Z = -\frac{m_{62}}{m_{22}}F_z$$
 (1-26)

where

$$M_{p} = -\frac{m_{62}^{m_{23}}}{m_{22}} + m_{63}$$
 (1-27)

$$M_{r} = -\frac{{}^{m}62^{m}24}{{}^{m}22}$$
 (1-28)

$$M_{u} = M_{53} - \frac{m_{51}^{m_{13}}}{m_{11}}$$
 (1-29)

$$M_{V} = -\frac{m_{51}^{m_{14}}}{m_{11}} \tag{1-30}$$

$$\left(m_{53} - \frac{m_{51}^{m} n_{13}}{m_{11}}\right) \ddot{\Theta}_{s} - \frac{m_{51}^{m} n_{14}}{m_{11}} \ddot{\Theta}_{i} + m_{55} \ddot{X} + m_{57} \ddot{n} + d_{55} \dot{X} + k_{55} \dot{X} = -\frac{m_{51}}{m_{11}} F_{x}$$
 (1-31)

$$m_{77}^{\ddot{\eta}} + d_{77}^{\dot{\eta}} + k_{77}^{\eta} = h_{zc}F_z + h_{xc}F_x$$
 (1-32)

These last five differential equations are rearranged so that a block diagram

can be constructed.

$$\ddot{\Theta}_{s} = -\frac{M_{i}}{M_{s}} \ddot{\Theta}_{i} - \frac{1}{M_{s}} \left(r_{cz} + \frac{m_{31}}{m_{11}} \right) F_{x} + \frac{1}{M_{s}} \left(r_{cx} - \frac{m_{32}}{m_{22}} \right) F_{z}$$

$$\ddot{\Theta}_{i} = -\frac{M_{k}}{M_{n}} \ddot{\Theta}_{s} - \frac{K_{r} \sigma_{rg}}{M_{n}} \dot{\eta} - \frac{k_{47}}{M_{n}} \eta - \frac{K_{i} \sigma_{ss}}{M_{n}} \int \eta dt + \frac{1}{M_{n}} T_{c}$$
(1-33)

$$-\frac{m_{41}}{M_{D}m_{11}}F_{x}-\frac{m_{42}}{M_{D}m_{11}}F_{z}$$
 (1-34)

$$\ddot{Z} = -\frac{M_p}{m_{66}} \ddot{\Theta}_s - \frac{M_r}{m_{66}} \ddot{\Theta}_i - \frac{m_{67}}{m_{66}} \ddot{\eta} - \frac{d_{66}}{m_{66}} \dot{Z} - \frac{k_{66}}{m_{66}} Z - \frac{m_{62}}{m_{22}m_{66}} F_z$$
 (1-35)

$$\ddot{X} = -\frac{M_u}{m_{55}} \ddot{\Theta}_s - \frac{M_v}{m_{55}} \ddot{\Theta}_i - \frac{m_{57}}{m_{55}} \ddot{\eta} - \frac{d_{55}}{m_{55}} \dot{x} - \frac{k_{55}}{m_{55}} x - \frac{m_{51}}{m_{11}m_{55}} F_x$$
 (1-36)

$$\ddot{\eta} = -\frac{d_{77}}{m_{77}} \dot{\eta} + \frac{k_{77}}{m_{77}} \eta + \frac{h_{zc}}{m_{77}} F_z + \frac{h_{xc}}{m_{77}} F_x$$
 (1-37)

The control torque T_c is given by Eq. (1-12).

The block diagram which portrays the differential equations of Eqs. (1-33) through (1-37) is shown in Fig. 1-1. The nonlinear element which represents the nonlinear torque due to the flex pivot and wire cable is also included in the block diagram. The signal flow graph representation of Fig. 1-1 for the purpose of evaluating the determinant is shown in Fig. 1-2. The dynamics of η is eliminated since they do not enter the determinant of the signal flow graph.

Applying Mason's gain formula to Fig. 1-2, the determinant of the signal flow graph is evaluated as follows:

$$\Delta = 1 + \frac{Ns + K_p s + K_i + K_r s^2 - \frac{M_v}{m_{55}} k_{45} G_x s^3 - \frac{M_r}{m_{66}} k_{46} G_z s^3}{M_n s^3}$$

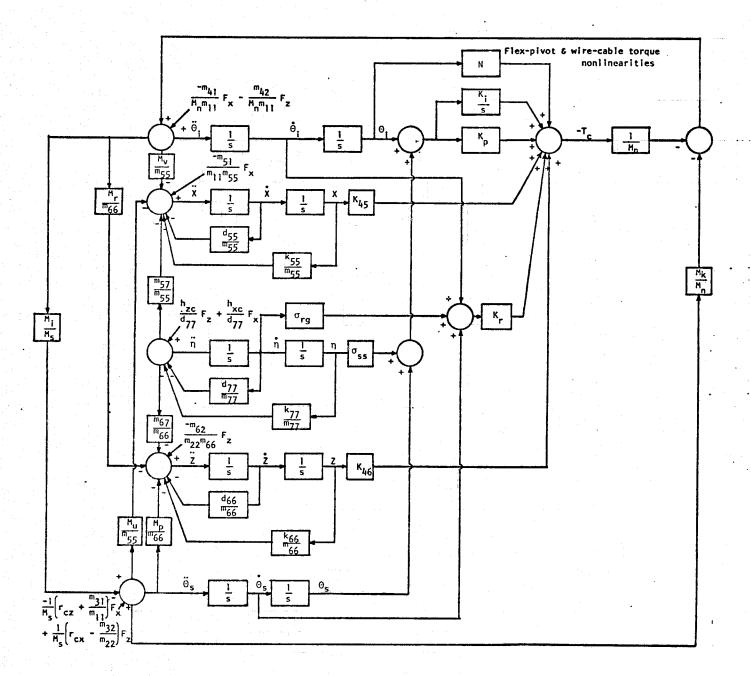


Figure 1-1. Block diagram of the IPS system with wire-cable and flex-pivot nonlinearities.

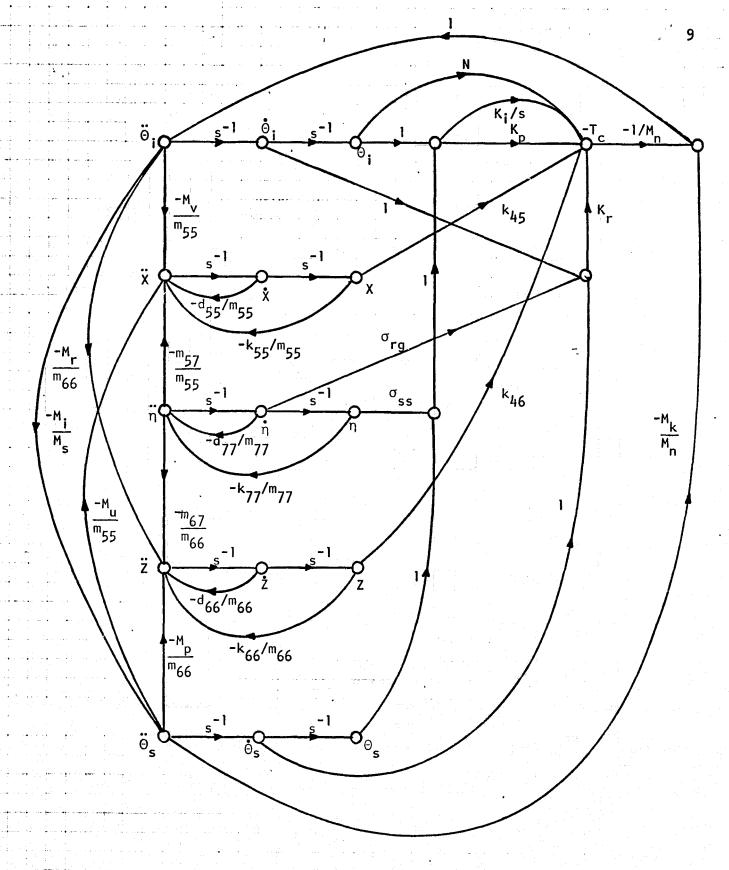


Figure 1-2. Signal flow graph of the IPS system,

+
$$\frac{M_{i} \left(\frac{M_{u}}{m_{55}} k_{45} G_{x} s^{3} + \frac{M_{b} k_{c}}{m_{66}} G_{z} s^{3} - K_{p} s - K_{i} - K_{r} s^{2} - M_{k} s^{3} \right)}{M_{s} M_{n} s^{3}} = 0 \quad (1-38)$$

where

$$G_{x} = \frac{1}{s^{2} + \frac{d_{55}}{m_{55}} s + \frac{k_{55}}{m_{55}}}$$
 (1-39)

$$G_{z} = \frac{1}{s^{2} + \frac{d_{66}}{m_{66}} s + \frac{k_{66}}{m_{66}}}$$
 (1-40)

Rearranging both sides of Eq. (1-38), we have

$$M_s M_n s^3 + M_s \left(Ns + K_p s + K_i + K_r s^2 - \frac{M_v}{m_{55}} k_{45} G_x s^3 - \frac{M_r}{m_{66}} k_{46} G_z s^3 \right)$$

+
$$M_{i} \left(\frac{M_{u}}{M_{55}} k_{45} G_{x} s^{3} + \frac{M_{p} k_{46}}{M_{66}} G_{z} s^{3} - K_{p} s - K_{i} - K_{r} s^{2} - M_{k} s^{3} \right) = 0$$
 (1-41)

Dividing both sides of the last equation by the terms that do not contain N, we get the equivalent linear transfer function that the nonlinear element N sees,

$$G_{eq}(s) = \frac{M_s s}{(M_s M_n - M_i M_k) s^3 + M_s (K_p s + K_i + K_r s^2 - \frac{M_v}{m_{55}} k_{45} G_x s^3 - \frac{M_r}{m_{66}} k_{46} G_z s^3)}$$

$$+ M_i \left(\frac{M_u}{m_{55}} k_{45} G_x s^3 + \frac{M_p k_{46}}{m_{66}} G_z s^3 - K_p s - K_i - K_r s^2\right)$$
(1-42)

For the system parameters given, $G_x = G_z$; thus, Eq. (1-42) is simplified to

$$G_{eq}(s) = \frac{M_s s \left(s^2 + \frac{d_{55}}{m_{55}} s + \frac{k_{55}}{m_{55}}\right)}{\left((M_s M_n - M_i M_k) s^3 + (M_s - M_i) (K_r s^2 + K_p s + K_i)\right) \left(s^2 + \frac{d_{55}}{m_{55}} s + \frac{k_{55}}{m_{55}}\right)}$$

$$\frac{1}{\left(\frac{M_i M_u - M_s M_v}{m_{55}}\right) k_{45} s^3 + \left(\frac{M_i M_p - M_s M_r}{m_{66}}\right) k_{46} s^3}$$
(1-43)

Substitution of the system parameters into Eq. (1-43) with

$$M_s = 7056977.95$$
 $M_k = M_i = 25992.12$
 $M_n = 10074.90$
 $M_u = 0.926522$
 $M_p = -4.537982$
 $M_r = 0.042982$
 $M_v = 0.02489$

we have

$$G_{eq}(s) = \frac{1.002077 \times 10^{-4} s (s^2 + 44s + 986)}{s^5 + 45.96676s^4 + 1107.4783s^3 + 2257.769s^2 + 7374.0808s + 10828.48765}$$
(1-44)

It is interesting to note that the system shown in Fig. 1-1 is of the llth order. However, since the dynamics of η do not enter the picture on the stability of the nonlinear system, and since $G_X=G_Z$, the transfer function seen by N is only of the 5th order.

II. PREDICTION OF SELF-SUSTAINED OSCILLATIONS OF THE CONTINUOUS-DATA IPS SYSTEM WITH WIRE-CABLE AND FLEX-PIVOT NONLINEARITIES

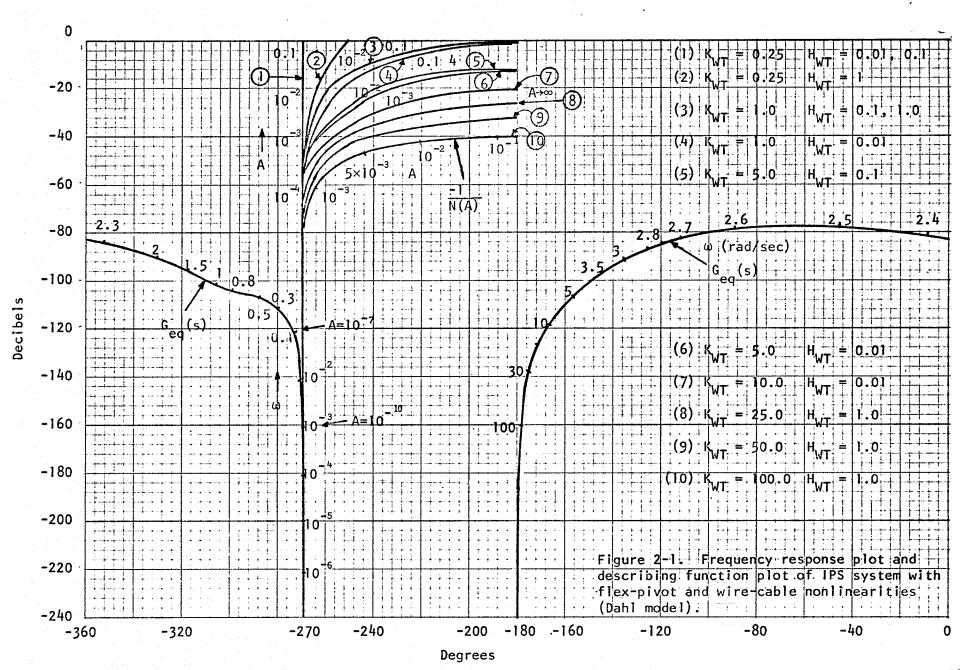
The equivalent transfer function $G_{eq}(s)$ in Eq. (1-44) can be plotted in the gain-phase coordinates together with the plot of -1/N of the nonlinear element for stability analysis. The describing function of the combined flex-pivot and wire-cable nonlinearity using the Dahl model has been derived in [1]. In this earlier report a simplified model of the IPS system was obtained by assuming that all but motion about two of the seven degrees of freedom of axes are negligible. Only motion about the scientific instrument axis and the mount rotation were considered.

For the purpose of comparison, the equivalent transfer function that the nonlinearity sees for the simplified IPS in [1] is repeated as follows:

$$G_{eq}(s) = \frac{0.0013946s(s^2 + 0.0012528s + 0.0036846)}{s^5 + 16.7s^4 + 222.6s^3 + 279.806s^2 + 5.857s + 5.13855}$$
(2-1)

when the integral control constant K_1 is 10^6 . The zeros of this transfer function are s=-0.0006264+j0.06069764 and s=-0.0006264-j0.06067674. The five poles are at s=-1.38135, s=-0.0031956+j0.135895, s=-0.0031956-j0.135895, s=-7.65618+j11.946, and s=-7.65618-j11.946. It was concluded in [1] from the gain-phase plot of $G_{eq}(s)$ versus the -1/N(A) plot that the simplified IPS may have a sustained oscillation that is characterized by the frequency of $\omega=0.16$ rad/sec. Depending on the values of the parameters of the wire-cable and flex-pivot nonlinearities, K_{wT} and H_{wT} , the amplitude of oscillation of ε , which is comparable to Θ_i in this report, lies between 3×10^{-8} to 3×10^{-6} for $K_1=10^6$.

The equivalent transfer function of Eq. (1-44) is plotted as shown in Fig. 2-1. It is noted that the IPS model used in the present work does not



have the same system parameters as the simplified IPS system studied in [1]. Thus, the natural frequencies of the two systems are different. From Fig. 2-1 it is observed that the $G_{\rm eq}(s)$ and -1/N loci would intersect only at very low frequencies and when the amplitude of oscillation is extremely small. For instance, at $\omega = 10^{-3}$ rad/sec, the amplitude of oscillation is approximately 10^{-10} , and for all practical purposes this can be regarded as zero. Thus, the IPS model considered here would not exhibit sustained oscillations due to the wire-cable and flex-pivot nonlinearities.

It is of interest to investigate the poles and zeros of the transfer function of Eq. (1-44). The poles and zeros of $G_{\rm eq}(s)$ of Eq. (1-44) are tabulated as follows:

zeros:
$$s = 0$$
, $-22 + j22.4053565$, $-22 - j22.4053565$

poles: $s = -1.66963$, $-0.123519 + j2.5232$
 $-0.123519 - j2.5232$
 $-22.025 + j23.0467$
 $-22.025 - j23.0467$

The natural frequency of the dominant poles of the $G_{\rm eq}(s)$ of Eq. (1-44) is found to be at 2.5232 rad/sec. This is much higher that that of the transfer function of Eq. (2-1). Furthermore, the two zeros listed above are very close to two of the poles.

NONLINEAR IPS CONTROL SYSTEM WITH DAHL MODEL AND ONE FLEXIBLE BODY MODE

The IPS control system with one flexible body mode and the nonlinear flex pivot torque modelled by the Dahl solid friction model (i = 2) is simulated on the digital computer. A block diagram of the IPS system is shown in Fig. 1-1. For the Dahl model with i = 2, the following parameters are used:

$$\gamma = 9.2444 \times 10^4$$

$$T_{FPO} = 2.25 \times 10^{-3} \text{ N-M}$$

The objective of the computer simulation of the system is to verify and/or clarify the analysis results obtained by the describing function method in Chapter II. Although the describing function method analysis in Chapter II does not yield precise results for possible amplitudes and frequencies of sustained oscillations, it does indicate that for the sets of system parameters used if sustained oscillations were to exist the amplitude of oscillation would be approximately less than 10⁻⁹ radians and the frequencies would be less than 10⁻³ radians/sec.

Since the significant time constant of the IPS system is very long, it would require a long computer simulation time to verify the existence or nonexistence of a limit cycle. Normally, the possibility of sustained oscillations could be checked by using two computer simulations, one with initial conditions that correspond to above and the other with initial conditions that correspond to below the value of the predicted amplitude of oscillations. If a sustained oscillation exists then the simulation with initial conditions below the predicted amplitude will result in a response that oscillates near the predicted frequency and increases in

amplitude. Also, the simulation with initial conditions that correspond to an initial amplitude above the predicted amplitude will result in a response that decreases in amplitude and oscillates near the predicted frequency. However, since the analysis of Chapter II does not predict any specific sustained oscillations, the method described above must be changed.

Since the analysis of Chapter II provides no specific candidates for amplitudes and frequencies of limit cycles (or the amplitude and frequencies are all extremely small), it may be necessary to check for the absence of limit cycles. The absence of limit cycles can be verified by showing that the responses of the system due to any initial conditions actually decay to zero as time increases indefinitely.

Again, since the time constant of the IPS system is very long, the simulation of the system response by a digital computer may involve excessive computer time. Since if sustained oscillations were to exist, the possible frequencies of oscillations would be less than 10^{-3} rad/sec, the simulation time required would be in excess of $2\pi \times 10^3$ sec. Also, a double-precision program with 30 digits of accuracy would be required, since the amplitude of oscillation must be less than 10^{-9} radians. Instead, the results from the simulation will be used to verify that the oscillations of the system are damped out at a mode that is practically independent of the nonlinearity.

As a means of exciting the system, an initial perturbation of $\Theta_i(0) = 10^{-9}$ radians was applied to the system. Two simulations of the system were completed for two extreme sets of wire cable parameters (K_{WT} and H_{WT}). The responses of Θ_i for both of these cases are shown in Figs. 3-1 and 3-2. These responses indicate that the influence of the wire cable parameters on the system response is very insignificant. In addition, the frequency of

```
Tim.
        THUTL
                                                                      THEOL
        1.0000-09
        5.0000E-01
                                                                     -2.3456-09
        -7.587E-10 +
1.00000100
                                                                     -1.2821. 09
1.50001400
        -7.140E+10 --+
                                                                      1.3835-07
        2.504E-10
7.663E-10
2.110L-10
2.00001.100
                                                                      1.9371-09
2.500001400
                                                                     -1.075E-10
-1.772E-09
3.000000100
       3.50000100
                                                                     -9.160E-10
4.000000100
                                                                      1.042E-07
4.5000E FOO
                                                                      1.4061.-07
        5.619E-10
5.0000E100
                                                                     -1.167E-10
        1.415E-10
5.5000E100
        -1.307E-09
6.0000F400
                                                                     -6.460E-10
7.877E-10
6.5000E+00
7.0000E+00
                                                                      1.022E-09
        4.112E-10
7.500001100
                                                                      -1.117E-10
        8.0000E+00
                                                                     -9.668E-10
8.5000E100
                                                                     -4.544E-10
       -2.676E-10 -----+
9.0000E100
                                                                      5.94DE-10
9.5000E100
                                                                      7.417E-10
1.00000E+01
                                                                     -1.011E-10
1.0500E#01
                                                                     -7.133E-10
1.100002401
                                                                     -3.188E-10
1.1500E+01
                                                                      4.481E-10
1.2000E+01
                                                                      5.381E-10
1.2500E+01
                                                                     -8.816E-11
1.3000E+01
                                                                     -5.260E-10
1.3500E+01
                                                                     -2.230E-10
1.40000+01
                                                                      3.373E-10
1.4500E+01
                                                                      3.901E-10
1.5000E+01
        2.352E-11 -----+
-1.323E-10 -----+
                                                                     -7.493E-11
1.5500E+01
                                                                     -3.876E-10
1.6000E+01
                                                                     -1.555E-10
        -1.013E-10 -----
1.6500E401
       2.536E-10
1.7000E+01
                                                                      2.82SE-10
1.75000401
                                                                     -6.247E-11
1.8000E+01
                                                                     -2.654E-10
1.8500E401
                                                                     -1.081F-10
1.9000E+01
                                                                      1.704E-10
1.9500E+01
                                                                      2.04UE-10
2.0000F+01
                                                                      -5.131E-11
2.0500E+01
       -2.101E-10
2.1000EF01
                                                                     -7.483E-11
2.1500E+01
                                                                      1.428E-10
2.2000F401
                                                                      1.479E-10
2.2500E+01
                                                                     -4.165E-11
2.30008+01
                                                                     -1.545E-10
2.3500E+01
                                                                     -5.159E-11
2.4000E+01
                                                                      1.070E-10
2.4500E+01
                                                         3-1.
                                                                      1.068E-10
        2.5000E+01
        -3.346E-11
2.5500E+01
                                                                     -1.136E-10
2.6000E+01
                                                                     -3.536E-11
2.6500E+01
        8.013E-11
                                                         Time
2.7000E+01
                                                      and
                                                                      7.7125-11
2.7500E+01
                                                                     -2.670E-11
        -1.763E-12 -----+
2.8000E+01
                                                                     -8.343E-11
2.8500E+01
        -2.413E-11
2.9000E+01
                                                         response
                                                                      5.993E-11
2.9500E+01
                                                                      5.5628-11
        3.0000E+01
                                                                     -2.11GE-11
3.0500E+01
                                                                     -6.125E-11
        -2.521E-11 -----+
-1.660E-11 -------
3.1000E+01
                                                                     -1.635E-11
3.1500E+01
                                                                      4.478E-11
4.007E-11
        7.939E-12
3.2000E401
        1.436E-11
3.2U00E+01
                                                                     -1.665E-11
                                                         of
        -3.648E-12
3.3000E+01
                                                                     -4.494E-11
        -1.963E-11
3.3500E+01
                                                                     -1.100E-11
        -1.286E-11 -----
                                                         0
3.4000E+01
                                                                      3.343E-11
        5.14SE-12 -----
3.4500E+01
                                                                      2.884E-11
                                                         Œ
        9.456E-12 ----
3.5000E+01
                                                                     -1.304E-11
        3.5500E+01
                                                                     -3.295E-11
3.6000E+01
                                                                     -7.327E-12
                                                         with
        3.6500E+01
                                                                      2.473E-11
3.7000E+01
                                                                      2.074E-11
3.7500E+01
        -4.148E-12
                                                                     -1.017E-11
3.8000E+01
        -2.415E-11
3.8500E+01
                                                                     -4.827E-12
3.9000E+01
        1.383E-12 -----
                                                                      1.8586-11
3.9500E+01
                                                                      1.4905-11
        3.26GE-12 ----
4.0000E+01
                                                                     -7.698E-12
        -4.203E-12
4.0500E+01
                                                                     -1.7588-11
        -1.014E-11 ----
4.1000E F01
        -3.137F-12
4.1500E+01
                                                                      1.3835-11
4.2000E101
       1.444E-13
1.363E-12
-4.197E-12
-0.459E-12
                                                                      1.069E-11
4.2500E101
                                                                     -6.112E-12
4.3000Ef01
                                                                     -1.294E-11
4.35000401
                                                                     -2.002E-12
        -5.97/E-12
4.4000EF01
                                                                      1.0270-11
        -7. Y63E-13 ----
4.45000101
                                                                      7.65GE-12
       -2.211E-14
4.5000E101
                                                                     -4.7155-12
```

. ,

oscillation is approximately 2.5 rad/sec, which is approximately 10⁴ times greater than the maximum possible frequency of oscillation. Also, the envelope of the response is decaying with a mode of σ = -0.1239, where 1/ σ is considered as the time constant. Hence, the system over the 50 sec interval of simulation time has a response due to a pair of poles at s = -0.1239 + j2.5 and s = -0.1239 - j2.5, which is approximately one of the pairs of poles of $G_{eq}(s)$. Thus, the system oscillates at the dominant mode of $G_{eq}(s)$ as expected, if the system had no limit cycles and the nonlinearity were replaced by a unity gain element. This implies that the nonlinearity has very little influence on the the system response under the simulated conditions, and that no limit cycles exist.

Since the size of the feasible amplitudes predicted is so small, at least 15 digits are needed to carry out the 50 seconds of computer simulation.

The listing of the computer simulation program is given in Table 3-1.

```
PROGRAM IPSSIM (INPUT, IPSD2, TAPE6=IPSD2, OUTPUT=IPSD2)
IMPLICIT REAL (M)
COMMON SIJSS, AKI, AKP, AKP, AK45, AK46, SIGPG, D77, M77, AK77, HZC, HXC
1, MT, MK, AS, MN, M41, M11, M31, RC7, M42, RCX, M32, M22, D55,
2AK55, M51, M55, M57, MU, MV, D66, M66, MP, MR, AK66, M67, M62,
3HWT, AKWT, R0, 31, SLAST, TFPO, II
EXTERNAL FCTY, OUTP
DIMENSION FCTY, OUTP
OINENSION CONSTANTS
AKWT=1.0E2, HWT=1.E0
GAMA=9.2444E4$TFPO=2.25E-3
G1=GAMA*TFPO
C INITIAL CONDITIONS
C INIPIAL CONDITIONS

Y (3) = 1. E-98SLAST=Y (3)

II=0

A=2. E0*G1*SLAST

A1=1. E0/A

R0=-1. E0*(-A1+SQRT(A1**2+1.E0))

C INIPIALIZE STATES
                                                                                                                                                                                                                        REPRODUCER
                          ORIGINAL PACE
           SÉI
         MASS
                             177=135

DAMPING CONSTANTS

D43=1.97E45D44=D43$D47=-76.4E0

D55=44.E05D66=D55$D77=.115E0

STIFFNESS CONSTANTS

AK43=7.E48AK44=AK43$AK45=-2.23E6

AK46=-3.87E65AK47=-2.72E2$AK55=9.86E2

AK66=9.35E25AK77=1.32E2
        SET
        SET
    AK46=-3.87=65AK47=-7.72E2#AKDD=9.00E2

AK66=9.36525AK77=1.32E2

SET PID CONSTANTS

AKR=D44#$3IG$G$=D47/AKR

AKR=AK44$5IG$G$=AK47/AKP

043=1.12581KI=043$Q44=Q43

SET DISTANCES FARM CENTER OF MASSES & CENTER OF GRAVITIES

SET DONE IF THE CREW MOTION IS NOT INCLUDED IN THE MODEL

4CC=1.E0$$RC=1.E0

RCY=1.E0$$RC=1.E0

BEFINE ARREADE ANS ELEMENTS

MI=M34-M31*M14/M11-M32*M24/M22

MS=M33-M13*M31/M11-M32*M23/M22

MS=M33-M13*M31/M11-M32*M23/M22

MS=M33-M13*M31/M11-M32*M23/M22

MS=M33-M13*M31/M11-M32*M24/M22

MS=M33-M13*M31/M11-M32*M24/M22

MS=M33-M51*M13/M11

EV=-M52*M24/M22

SET GROR WEIFHTING FACTORS

DO 5 I=1,ND14

DO 5 I=1,ND14

SET GROR WEIFHTING FACTORS

DO 5 I=1,ND14

PRMT(2)=0.50

PRMT(3)=1.E-2

PRMT(3)=1.E-2

PRMT(4)=1.E-14

PRMT(5)=0.E0
                              PRMT
PRMT
PRMT
                                                  (4) = 1. E-14
(5) = 3. E 7
(6) = 2.5E-1
(7) = PR 4 F (1)
RKGS (PR 4 T, Y, DERY, NDIN, IHLF, FCTY, OUTP, LUX)
                               PRMT
                               CALL
                               STOP
```

END

```
SUBROUTINE OUTP (TIME, Y, DERY, IHLF, NDIM, PRMT)
DIMENSION Y (NDIM), DERY (NDIM), PRMT (7)
DEL=TIME-PRMT (7)
IF (ABS (TIME-0.E0) .LE. 1.E-10) GO TO 1
IF (ABS (PRMT (5) - DEL) .LE. 1.E-10) GO TO 1
RETURN
                          PRITE (7) = TIME
WRITE (5, 10) IHLF, TIME, Y (3), Y (8)
FORMAT (16, 1PE10.4, 1P2E11.5)
                            RETURN
                            END
                          SUBROUTINE FCTY (TIME, STATE, STDOT, KEEP, NDIM)
IMPLICIT REAL (M)
COMMON SIGSS, AKI, AKP, AKR, AK45, AK46, SIGRG, D7
IMPLICIT REAL (M)

COMMON SIGSS, AKI, AKP, AKR, AK45, AK46, SIGRG, D77, M77, AK77, HZC

1, HXC, 41, MK, M5, M1, M41, M11, M31, PCZ, M42, RCX, M32, M22, D55

2, AK55, M51, 455, M57, MU, MV, D66, M66, NP, MR, AK66, M67, M62, HWT,

3AKWT, BC, G1, SLAST, TF BC, II

THIS SUBROUTINE ONLY UPDATES THE DEPIVATIVE OF THE STATE VECTOR

THE UPDATE DEPENDS ON THE IMPUTS, TIME, STATE, & KEEP.

DIMENSION STATE (NDIM), STDOT (NDIM)

FX (TIME) = 0. E0

10 FORMAT (16, 1P210.4, 1P2E11.3)

MODEL THE WIRE NONLINEARITY

SGNEDT=1. E0

IF (STATE(3).LT.O.EO) SGNEDT=-1.E0

INC=SGNEDT*HHIT*AKWT*STATE(3)

MODEL FOR THE DAHL NONLINEARITY (I=2)

IF (TIME.GT.J.EO).AND. (KEEP.EQ.O)) GO TO 1

GO TO 2

1 II=1
                                                                                                                                                                                                                                                                                                 THE STATE VECTOR.
                            Rī=RO-SGNSDI
                          RT=RO-SSHEDT

TFP=(SGHEDT+R1/(1.EG-G1*(STATE(3)-SLAST)*R1))*TFPC

IF (TFP.GT.TF20) TFP=TFPO

IF (TFP.LT.-TFPO) TFP=-TFPO

IF (II.EQ.O) GO TO 3

RO=TFP/TFPO
                            II=0
                            SLAST=STATE (3)
                 SLAST=STATE(3)

GO TO:

TFP=3.F)*TFP

ENN=TFP+TWC
STDOT(1) = STATE(7)
STDOT(2) = STATE(7)
STDOT(3) = STATE(3)
STDOT(4) = STATE(3)
STDOT(4) = STATE(3)
STDOT(1) = STATE(3)
STDOT(1) = STATE(1)
FCM=ENN+AKI*STATE(1) + SKP*STDOT(11) + AK45*STATE(4) + AK46*STATE(
15) + AK8*(STATE(1) + STATE(5) - AK77/M7*STATE(1)

STDOT(6) = -D.77/M77*STATE(5) - AK77/M77*STATE(1)

1+HCC/D77*P2(TIME) + FY (TIME) + HCC/D77
STDOT(7) = 1. E)/(1. EO-MI*MK/MS/MN) * (LI/MS/MN*TCM+(MI*M41/(MS*MN*
1411) - (SC2+M31/M1)/MS/*FY (TIME) + (NI*M42/(MS*MN*M11) + (PCX

STDOT(8) = -1K/MN*STDOT(7) - TCM/MN-M41/MN/M11*FX (TIME) - M42/MN/
STDOT(8) = -1K/MN*STDOT(7) - TCM/MN-M41/MN/M11*FX (TIME) - M42/MN/
STDOT(9) = -D.55/M55*STATE(9) - AK55/M55*STATE(4) - M51/M11/M55*FX (
IIME) - M57/455*STDOT(6) - MU/M55*STDOT(7) - NV/M55*STDOT(8)
STDOT(10) = -D.56/M66*STATE(10) - AK66/M66*STATE(5) - M67/M66*STDOT(
KEEP=1
RETURN
END
                           GO TO
        3
                          END
```

```
Table 3-1 (continued).
```

```
SUBROUTINE RKGS (PRMT, Y, DERY, NDIM, IHLF, FCT, OUTP, AUX)
CC.
          CALL FCT (X, Y, DERY, KEEP, NDIM)
C
          ZRROR TEST
          IF (H* (XEND-X)) 38,37,2
CC
          PREPARATIONS FOR RUNGE-KUTTA METHOD
         A (1) = .5
A (2) = .2928932'
A (3) = 1.707127
A (4) = .1666667
3 (1) = 2.
       2
            \begin{cases} 2 \\ 3 \\ = 1 \end{cases}
                                                                              REPRODUCIPILATE OF T
          BC 241 = .5
                                                                              ORIGINAL PART IN MY
          2 \ 1 \ = .5

2 \ 2 \ = .2923932

2 \ 3 \ = 1.707107

3 \ 4 \ = .5
          PREPARATIONS OF

DO 3 I=1, NDIM

AUX (1, I) = Y (I)

AUX (2, I) = DERY (I)

AUX (3, I) = 0.

AUX (6, I) = 0.

INCLEDED
                                    FIRST PUNGE-KUTTA STEP
          H = H + H
          IHLY= -1
ISTEP=0
          IEND=0
CCC
          START OF A RUNGE-KUTTA STEP
IF ((X+H-XEND)*H)7,6,5
H=XEND-X
       6 IEND=1
          RECORDING OF INITIAL VALUES OF THIS STEP CALL OUTP(K,Y,DERY,IREC,NDIM,PRMT) IF (PRMC(1+4)) 40,3,40 ITEST=0
          ISTEP=ISTEP+1
          START OF INNERMOST PUNGF-KUTTA LOOP J=1
    IF (J-3) 13, 14, 13

X=X+.5*H

CALL FCT (X,Y,DERY,KEEP,NDIM)

GOTO 10
      14
          GOTO
CC
           END OF INNERMOST RUNGE-KUTTA LOOP
```

```
Č.
         TEST OF ACCURACY
15 IF (ITEST) 16,16,29
                  IN CASE ITEST=0 THERE IS NO POSSIBILITY FOR TESTING OF ACCURACY DO 17 I=1, NDIM

AUX (4, T) = Y(I)

ITEST=1

ISTEP=ISTEP+ISTEP-2

IHLF=IHLF+1
          15
17
          13
                 Int = 1 H L F + 1

X = X - H

H = . 5 * H

DO 19 I = 1, VOIM

Y(I) = AUX (1, I)

DERY (I) = AUX (2, I)

AUX (5, I) = AUX (3, I)

GOTO 5
                IN CASE ITEST=1 TESTING OF ACCURACY IS POSSIBLE IMOD=ISTEP/2
IF (ISTEP-IMOD-IMOD) 21,23,21
CALL FCT (X,Y,DERY,KEFP,NDIM)
DO 22 I=1,NDIM
AUX (2,I) = Y(I)
AUX (7,I) = DERY (I)
GOTO 9
          2)
CC
                  COMPUTATION OF TEST VALUE DELT
                 DELT=0.
DO 24 I=1,NDIM
DELT=DELT+AUX (8,I) *ABS (AUX (4,I) -Y(I))
IF (DELT-PRMT (1+3)) 28,29,25
          23
                 ERROR IS TOO GREAT
IF (IHLE-1), 26, 36, 36
DO 27 I=1, UDIM
AUX (4, I) = AUX (5, I)
ISTEP=ISTEP+ISTEP-4
X=X-H
                  Ï EÑ D≌ O
                  GOTO 18
CC
                  RESULT VALUES ARE GOOD
                KEEP=0

CALL FCT(X,Y,DERY,KEEP,NDIM)

DO 29 I=1,NDIM

AUX (1,I) = Y(I)

AUX (2,I) = DERY (I)

AUX (3,I) = AUX (6,I)

Y(I) = AUX (5,I)

DERY (I) = AUX (7,I)

CALL OUTP(X-I,Y,DERY,IHLF,NDIM,PRMT)

IF (PRMT (1+4)) 40,30,45

DO 31 I=1,NDIM

Y(I) = AUX (1,I)

DERY (I) = AUX (2,I)

IREC=IHLF

IF (KEEP-EJ. 2) GO TO 39

IF (IEND) 32,32,39
   23
        INCREMENT GETS DOUBLED

15TEP=ISTEP/2
                  H = \Pi + H
        ISTEP=TSTEP/2
                 H=H+H
                 GOTO 4
C
```

```
FORMAT REQUIRED:
FORMAT (6X, 17E10. 4, 1P10E11.3)
6X = THIS FIELD WIDTH CAN BE USED TO STORE THE OR SOME OTHER PARM.
1PE10.4 = THIS FIELD IS USED TO STORE THE INDEPENDENT VARIABLE (TIME)
1P10E11.3 = THESE FIELD WIDTHS CONTAIN UP TO 10 COLS. OF DATA
           IF (WAIT.NT."W") GO TO 1001
PRINT 1002
                      "TYPE CARRIAGE RETURN TO BEGIN OUTPUT")
READ (5,1000) (CARD (L), L=1,80)
FORMAT (80A1)
ISKIP=30000
TTYNUM=73.0
           1002
                                                                                         CHARACTERS ?"./.
           1000
1001
                      ĬŔŢĸŨĬŦĊŶĹĬQ. 132) GO TO 10
ŦŦŶŊŨŊ=55.0
                      NUMPRT=0

ANAME="TIME"
MAXNUM=0
           10
                      MAX GOL=HUMCOL
MAXCOL=HUMCOL
IF (NUMP1.GT. MAXCOL) MAXCOL=HUMP1
IF (NUMP2.GT. MAXCOL) MAXCOL=HUMP2
READ (6, 30
IF (EOF (5) . NE. 0) GO TO 100
FORMAT (6X, G10.4, 10G11.3)
           2)
           30
```

```
100
110
120
130
4)
50
60 .
   IZ001=K-1

DO 70 N=1, IZ001

CSTORE(N)="-"

IZ002=K+1

DO 77 H-17002
70
75
    77
9)
92
89
```

IV. DISCRETE DESCRIBING FUNCTION OF THE COMBINED WIRE-CABLE TORQUE AND FLEX-PIVOT NONLINEARITY

In the last three chapters the continuous-data IPS system with the combined nonlinearity of the wire-cable torque and flex pivot is studied.

The describing function analysis is applied to the continuous-data system.

In the present chapter the digital IPS system with the combined nonlinearity is considered. The block diagram of the system is shown in Fig. 4-1. The combined nonlinearity is represented by the block N. The digital system is characterized by having the sample-and-hold units which have the notation S/H. In order to apply the discrete describing function technique, a sample-and-hold unit is inserted in the path of the nonlinearity N.

The objective of the investigation is to study whether self-sustained oscillations will occur in the digital IPS system. If self-sustained oscillations do occur, as they most likely will in a digital nonlinear system, what are the amplitudes and frequencies of these oscillations?

4.1 Discrete Describing Function of the Combined Nonlinearity

The discrete model of the combined nonlinearity of the wire-cable torque and the flex-pivot characteristics is shown in Fig. 4-2. The zero-order hold at the input of the combined nonlinearity is deleted, since there is already a zero-order hold at the output of N.

The wire-cable torque $T_{\rm wc}$ is modelled as shown in Fig. 4-2 and is functionally represented by

$$T_{WC} = H_{WT} SGN(\dot{\Theta}_{i}) + K_{WT} \Theta_{i}$$
 (4-1)

or

26

$$T_{WC}^{+}(\Theta_{i}) = H_{WT} + K_{WT}\Theta_{i} \qquad \qquad \dot{\Theta}_{i} \geq 0 \qquad (4-2)$$

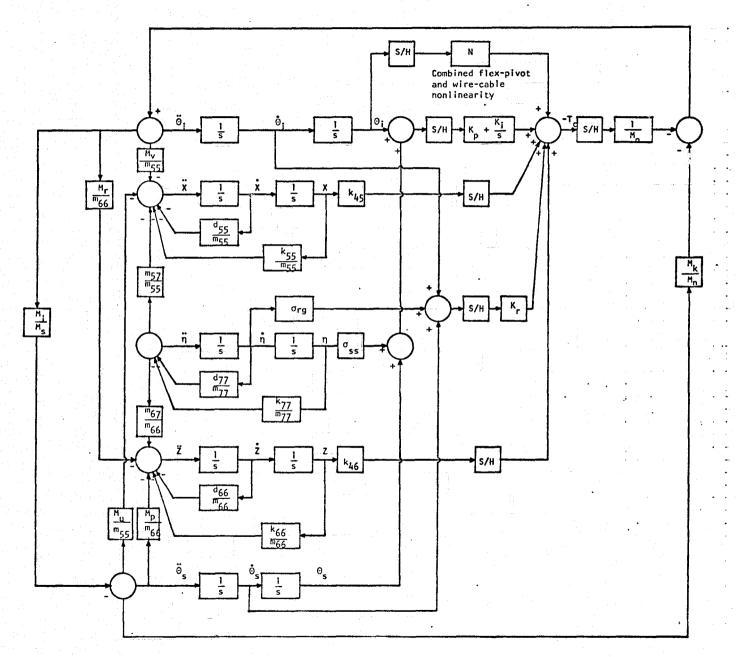


Figure 4-1. Block diagram of the digital IPS system with flex-pivot and wire-cable nonlinearities.

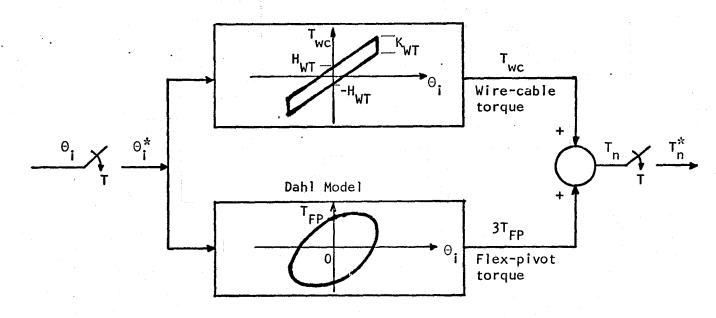


Figure 4-2. Discrete model of the combined nonlinearity of the wire-cable and flex-pivot torques.

$$T_{WC}(\Theta_i) = -H_{WT} + K_{WT}\Theta_i \qquad \qquad \dot{\Theta}_i \leq 0 \qquad (4-3)$$

where H_{WT} is in N-m, K_{WT} in N-m/rad. Θ_{i} is in rad, and $T_{WC}(\Theta_{i})$ in N-m.

It has been established that the Dahl solid rolling friction characteristics can be approximated by the nonlinear relation,

$$\frac{dT_{FP}(\Theta_i)}{d\Theta_i} = \gamma (T_{FPI} - T_{FPO})^i$$
 (4-4)

where

i = positive number

 γ = positive constant

$$T_{FPI} = T_{FP} SGN(\dot{\Theta}_{i})$$

 T_{FPO} = saturation level of T_{FP}

For i = 2, Eq. (4-4) is integrated to give

$$\Theta_{i} + C_{1} = \frac{-1}{\gamma(T_{EP}^{+} - T_{EP0})}$$
 $\Theta_{i} \ge 0$ (4-5)

$$\Theta_{i} + C_{2} = \frac{-1}{\gamma(T_{FP}^{-} + T_{FPO}^{-})} \qquad \dot{\Theta}_{i} \leq 0$$
 (4-6)

* --- -

. where $\mathbf{C_1}$ and $\mathbf{C_2}$ are constants of integration, and

$$T_{FP}^{+} = T_{FP} \qquad \qquad \Theta_{i} \geq 0 \qquad (4-7)$$

$$T_{FP}^- = T_{FP}^- \qquad \qquad \Theta_i \leq 0 \qquad (4-8)$$

The constants of integration are determined at the initial point where

$$\Theta_{ii}$$
 = initial value of Θ_{i}
 T_{FPi} = initial value of T_{FP}

Then,

$$c_1 = -\Theta_{ii} - \frac{1}{\gamma(T_{FPi}^+ - T_{FP0}^-)}$$
 $\dot{\Theta}_i \ge 0$ (4-9)

$$c_2 = -\Theta_{ii} - \frac{1}{\gamma(T_{FPi}^+ + T_{FPO}^-)}$$
 $\Theta_i \le 0$ (4-10)

The describing function analysis depends on the assumption that the input to the nonlinearity is a sine wave. Let $\theta_i(t)$ be described by the cosinusoidal function,

$$\Theta_{i}(t) = A\cos\omega t$$
 (4-11)

Then,
$$\Theta_{i}(t) = -A\omega \sin \omega t$$
 (4-12)

Thus,
$$\Theta_{ii} = -A$$
 $\dot{\Theta}_{i} \geq 0$ (4-13)

$$\Theta_{i,j} = A \qquad \qquad \Theta_{i,j} \leq 0 \qquad (4-14)$$

The constant of integration in Eqs. (4-9) and (4-10) become

$$c_1 = A - \frac{1}{\gamma(T_{FPi}^+ - T_{FP0}^-)}$$
 (4-15)

$$c_2 = -A - \frac{1}{\gamma(T_{FPi}^+ - T_{FP0}^-)}$$
 (4-16)

Substitution of Eqs. (4-11) and (4-15) in Eq. (4-5) and simplifying, the solution of T_{FP}^+ is written as

$$T_{FP}^{+}(t) = T_{FP0} \frac{\frac{R}{R-1} + \frac{a}{2}(1 - \cos\omega t)}{\frac{a}{2}(1 - \cos\omega t) + \frac{1}{R-1}}$$
(4-17)

which is valid for $\dot{\Theta}_i \geq 0$ or $(2k+1)\pi \leq \omega t \leq 2(k+1)\pi$, $k=0,1,2,\ldots$, where

$$a = 2\gamma AT_{FPO} \tag{4-18}$$

$$R = -\frac{1}{a} + \sqrt{\frac{a^2 + 1}{a^2}} = \frac{T_{FPi}}{T_{FP0}}$$
 (4-19)

Similarly, for $\dot{\Theta}_{i} \leq 0$, using Eqs. (4-11) and (4-16) in Eq. (4-6), we have

$$T_{FP}^{-} = T_{FPO} \frac{\frac{R}{R+1} - \frac{a}{2}(1 - \cos\omega t)}{\frac{a}{2}(1 - \cos\omega t) + \frac{1}{R+1}}$$
(4-20)

which is valid for $2k\pi \le \omega t \le (2k+1)\pi$, $k=0,1,2,\ldots$

The relations for T_{FP}^+ and T_{FP}^- in Eqs. (4-17) and (4-20) together with those of $T_{WC}(\Theta_i)$ in Eqs. (4-2) and (4-3) are used for the derivation of the discrete describing function of the combined nonlinearity. As shown in Fig. 4-2, the total torque disturbance due to the combined nonlinearity is given by

$$T_{n}^{+} = T_{WC}^{+} + 3T_{FP}^{+}$$

$$= H_{WT} + K_{WT}^{-}A\cos\omega t + 3T_{FP0} \frac{\frac{R}{R-1} + \frac{a}{2}(1 - \cos\omega t)}{\frac{a}{2}(1 - \cos\omega t) + \frac{1}{R-1}} \quad \Theta_{i} \geq 0 \quad (4-21)$$

$$T_{\overline{n}}^{-} = T_{WC}^{-} + 3T_{FP}^{-}$$

$$= -H_{WT} + K_{WT}^{-}A\cos\omega t + 3T_{FPO} \frac{\frac{R}{R+1} - \frac{a}{2}(1 - \cos\omega t)}{\frac{a}{2}(1 - \cos\omega t) + \frac{1}{R+1}} \quad \dot{\Theta}_{i} \leq 0 \quad (4-22)$$

For the discrete model, we let

$$\Theta_{t}(t) = A\cos(\omega t + \phi) \qquad (4-23)$$

where ϕ denotes the phase in radians. The z-transform of Θ_i (t) is

$$\Theta_{i}(z) = \sum_{k=0}^{\infty} A\cos\left(\frac{2\pi k}{N} + \phi\right) z^{-k}$$
 (4-24)

 $N = 2,3,\ldots$ Or, in closed form,

$$\Theta_{\mathbf{i}}(z) = \frac{Az((z - \cos 2\pi/N)\cos \phi - \sin 2\pi/N \cdot \sin \phi)}{z^2 - 2z\cos 2\pi/N + 1}$$
(4-25)

The z-transform of $T_n(t)$ is denoted by $T_n(z)$. Then, the discrete describing function of the combined nonlinearity is defined as

$$N(z) = \frac{T_n(z)}{\Theta_i(z)}$$
 (4-26)

The discrete describing function (DDF) for N = 2 is derived separately in the following section.

4.2 The DDF of the Combined Nonlinearity For N = 2

Let $T_n(kT)$ denote the value of $T_n^*(t)$ at t=kT. For N=2, the signal $T_n^*(t)$ is a periodic function with a period of 2T. The z-transform of $T_n^*(t)$ is written as

$$T_{n}(z) = T_{n}(0)(1 + z^{-2} + z^{-4} + \dots) + T_{n}(T)(z^{-1} + z^{-3} + \dots)$$

$$= \frac{T_{n}(0)z^{2} + T_{n}(T)z}{z^{2} - 1}$$
(4-27)

For the cosinusoidal input of Eq. (4-23), the corresponding expression for $T_{FP}^+(t)$ and $T_{FP}^-(t)$ are

$$T_{FP}^{+}(t) = T_{FP0} \frac{\frac{R}{R-1} + \frac{a}{2}(1 - \cos(\omega t + \phi))}{\frac{1}{R-1} + \frac{a}{2}(1 - \cos(\omega t + \phi))} \qquad \dot{\theta}_{i} \geq 0 \qquad (4-28)$$

$$T_{FP}^{-}(t) = T_{FP0} \frac{\frac{R}{R+1} - \frac{a}{2}(1 - \cos(\omega t + \phi))}{\frac{1}{R+1} + \frac{a}{2}(1 - \cos(\omega t + \phi))} \qquad \dot{\theta}_{i} \leq 0 \qquad (4-29)$$

respectively. For t = kT, the last two equations become

$$T_{FP}^{+}(kT) = F_{FP0} \frac{\frac{R}{R-1} + \frac{a}{2}(1 - \cos(\frac{2\pi k}{N} + \phi))}{\frac{1}{R-1} + \frac{a}{2}(1 - \cos(\frac{2\pi k}{N} + \phi))} \qquad \theta_{i} \geq 0 \qquad (4-30)$$

$$T_{FP}^{-}(kT) = T_{FP0} \frac{\frac{R}{R+1} - \frac{a}{2}(1 - \cos(\frac{2\pi k}{N} + \phi))}{\frac{1}{R+1} + \frac{a}{2}(1 - \cos(\frac{2\pi k}{N} + \phi))} \qquad \theta_{i} \leq 0 \qquad (4-31)$$

For N = 2, Eq. (4-25) is simplified to

$$\Theta_{i}(z) = \frac{Az\cos\phi}{z+1} \tag{4-32}$$

Substitution of Eqs. (4-32) and (4-27) into Eq. (4-26), we have

$$N(z) = \frac{T_n(z)}{\Theta_1(z)} = \frac{T_n(0)z + T_n(T)}{A(z - 1)\cos\phi}$$
 (4-33)

Also, for N = 2, z = -1; the last equation is simplified to

$$N(z) = \frac{T_{n}(0) - T_{n}(T)}{2A\cos\phi}$$
 (4-34)

For N = 2,
$$T_{n}(0) = T_{n}^{-}(0) \qquad 0 \leq \phi < \pi \ (\stackrel{\circ}{\Theta}_{i} \geq 0)$$

$$= T_{n}^{+}(0) \qquad \pi \leq \phi < 2\pi \ (\stackrel{\circ}{\Theta}_{i} \leq 0)$$

$$T_{n}(T) = T_{n}^{+}(T) \qquad 0 \leq \phi < \pi \ (\stackrel{\circ}{\Theta}_{i} \geq 0)$$

$$= T_{n}^{-}(T) \qquad \pi \leq \phi < 2\pi \ (\stackrel{\circ}{\Theta}_{i} \leq 0)$$

Then, Eq. (4-34) becomes

$$N(z) = \frac{T_n^-(0) - T_n^+(T)}{2A\cos\phi} \qquad 0 \le \phi < \pi \qquad (4-35)$$

$$N(z) = \frac{T_n^+(0) - T_n^-(T)}{2A\cos\phi} \qquad \pi \le \phi < 2\pi$$
 (4-36)

For stability analysis, it is convenient to define

$$F(z) = -\frac{1}{N(z)}$$
 (4-37)

Using Eqs. (4-2), (4-3), (4-30) and (4-31), we have

$$T_{n}^{+}(0) = H_{WT} + K_{WT}^{A\cos\phi} + 3T_{FPO} \frac{\frac{R}{R-1} + \frac{a}{2}(1 - \cos\phi)}{\frac{1}{R-1} + \frac{a}{2}(1 - \cos\phi)}$$
(4-38)

$$T_{n}^{-}(0) = -H_{WT} + K_{WT}^{-}A\cos\phi + 3T_{FP0} \frac{\frac{R}{R+1} - \frac{a}{2}(1 - \cos\phi)}{\frac{1}{R-1} + \frac{a}{2}(1 - \cos\phi)}$$
(4-39)

$$T_{n}^{+}(T) = H_{WT} + K_{WT}A\cos(\phi + \pi) + 3T_{FPO} \frac{\frac{R}{R-1} + \frac{a}{2}(1 - \cos(\pi + \phi))}{\frac{1}{R-1} + \frac{a}{2}(1 - \cos(\pi + \phi))}$$

$$= H_{WT} - K_{WT}A\cos\phi + 3T_{FPO} \frac{\frac{R}{R-1} + \frac{a}{2}(1 + \cos\phi)}{\frac{1}{R-1} + \frac{a}{2}(1 + \cos\phi)}$$
(4-40)

$$T_{n}^{-}(T) = -H_{WT} - K_{WT}A\cos\phi + 3T_{FPO} \frac{\frac{R}{R+1} - \frac{a}{2}(1 + \cos\phi)}{\frac{1}{R+1} + \frac{a}{2}(1 + \cos\phi)}$$
(4-41)

Thus, for $0 \le \phi < \pi$,

$$T_{c}^{-}(0) - T_{c}^{+}(T) = 2K_{WT}^{-}A\cos\phi - 2H_{WT} + 3T_{FPO} \left(\frac{\frac{R}{R+1} - \frac{a}{2}(1 - \cos\phi)}{\frac{1}{R+1} + \frac{a}{2}(1 - \cos\phi)} - \frac{\frac{R}{R-1} + \frac{a}{2}(1 + \cos\phi)}{\frac{1}{R+1} + \frac{a}{2}(1 - \cos\phi)} \right)$$
(4-42)

For $\pi \leq \phi < 2\pi$,

$$T_c^+(0) - T_c^-(T) = 2K_{WT}A\cos\phi + 2H_{WT} + 3T_{FPO} \left[\frac{\frac{R}{R-1} + \frac{a}{2}(1 - \cos\phi)}{\frac{1}{R-1} + \frac{a}{2}(1 - \cos\phi)} \right]$$

$$-\frac{\frac{R}{R+1} - \frac{a}{2}(1 + \cos\phi)}{\frac{1}{R+1} + \frac{a}{2}(1 + \cos\phi)}$$
 (4-43)

4.3 Properties of F(z) = -1/N(z) for N = 2 as $A \rightarrow 0$ and $A \rightarrow \infty$

The properties of F(z) for N=2 as $A \to 0$ and $A \to \infty$ are now investigated. These properties will be useful in the determination of the critical regions of F(z) for stability studies.

Theorem 4-1.

For N = 2,

$$\lim_{A\to\infty} F(z) = \lim_{A\to\infty} -1/N(z) = -1/K_{WT} \quad \text{for all } \phi \qquad (4-44)$$

Proof: From Eq. (4-19),

$$\frac{R}{R-1} = \frac{-1 + \sqrt{a^2 + 1}}{-1 - a + \sqrt{a^2 + 1}}$$
 (4-45)

$$\frac{R}{R+1} = \frac{-1 + \sqrt{a^2 + 1}}{-1 + a + \sqrt{a^2 + 1}}$$
 (4-46)

$$\frac{1}{R-1} = \frac{a}{-1-a+\sqrt{a^2+1}} \tag{4-47}$$

$$\frac{1}{R+1} = \frac{a}{-1+a+\sqrt{a^2+1}} \tag{4-48}$$

and $a = 2\gamma AT_{FPO}$.

Let $F^+(z) = F(z)$ for $0 \le \phi < \pi$. Substituting Eq. (4-35) into Eq. (4-37), we have

$$F^{+}(z) = \frac{-2A\cos\phi}{T_{n}^{-}(0) - T_{n}^{+}(T)}$$
 (4-49)

Then,

$$\lim_{A \to \infty} F^{+}(z) = \lim_{A \to \infty} \frac{-2A\cos\phi}{T_{n}^{-}(0) - T_{n}^{+}(T)}$$
(4-50)

Substituting Eqs. (4-39) and (4-40) into the last equation and using Eqs. (4-45) through (4-48), we get

$$\lim_{A\to\infty} F^+(z) =$$

$$\begin{array}{c} -2A\cos\phi \\ \\ A \rightarrow \infty \end{array}$$

$$\begin{array}{c} -1 + \sqrt{a^2 + 1} - \frac{a}{2}(1 - \cos\phi) - \frac{-1 + \sqrt{a^2 + 1}}{-1 - a + \sqrt{a^2 + 1}} + \frac{a}{2}(1 + \cos\phi) \\ \\ -1 + a + \sqrt{a^2 + 1} + \frac{a}{2}(1 - \cos\phi) - \frac{a}{-1 - a + \sqrt{a^2 + 1}} + \frac{a}{2}(1 + \cos\phi) \end{array} + \\ \end{array}$$

$$= \lim_{A \to \infty} \frac{-2A\cos\phi}{-6T_{\text{FPO}} + 2K_{\text{WT}}A\cos\phi - 2H_{\text{WT}}} = \lim_{A \to \infty} \frac{-A\cos\phi}{K_{\text{WT}}A\cos\phi} = -\frac{1}{K_{\text{WT}}} \quad \text{Q.E.D.}$$

Similarly, for $\pi \leq \phi < 2\pi$, $F^{-}(z) = F(z)$,

$$\lim_{A\to\infty} F^{-}(z) = \lim_{A\to\infty} \frac{-2A\cos\phi}{T_n^{+}(0) - T_n^{-}(T)}$$

$$\begin{array}{c}
-2A\cos\phi \\
3T_{\text{FPO}} = 1 \text{ im} \\
\frac{-1 + \sqrt{a^2 + 1}}{a} + \frac{a}{2}(1 - \cos\phi) - \frac{-1 + \sqrt{a^2 + 1}}{-1 + a + \sqrt{a^2 + 1}} - \frac{a}{2}(1 + \cos\phi) \\
\frac{a}{-1 + a + \sqrt{a^2 + 1}} - \frac{a}{2}(1 - \cos\phi) - \frac{-1 + \sqrt{a^2 + 1}}{-1 + a + \sqrt{a^2 + 1}} + \frac{a}{2}(1 + \cos\phi) \\
+ \frac{a}{-1 + a + \sqrt{a^2 + 1}} + \frac{a}{2}(1 + \cos\phi)
\end{array}$$

$$= \lim_{A \to \infty} \frac{-2A\cos\phi}{3T_{\text{FPO}} \left[\frac{-a + \frac{a}{2}(1 - \cos\phi)}{\frac{1}{2} + \frac{a}{2}(1 - \cos\phi)} - \frac{\frac{1}{2} - \frac{a}{2}(1 - \cos\phi)}{\frac{1}{2} + \frac{a}{2}(1 + \cos\phi)} \right] + 2K_{\text{WT}}A\cos\phi + 2H_{\text{WT}}$$

$$= \lim_{A \to \infty} \frac{-2A\cos\phi}{6T_{\text{FPO}} + 2K_{\text{WT}}A\cos\phi + 2H_{\text{WT}}} = -\frac{1}{K_{\text{WT}}}$$
 Q.E.D.

Theorem 4-2.

For N = 2,

$$\lim_{A \to 0} F(z) = \lim_{A \to 0} -\frac{1}{N(z)} = 0/180^{\circ} + \tan^{-1} \frac{\cos \phi}{|\cos \phi|} \qquad \text{for all } \phi \qquad (4-51)$$

Proof: For $0 < \phi < \pi$,

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$$\lim_{A\to 0} F^+(z) =$$

$$= \lim_{A \to 0} \frac{-2A\cos\phi}{3T_{\text{FPO}} \left[\frac{\frac{a}{1+a} - \frac{a}{2}(1-\cos\phi)}{\frac{1}{1+a} + \frac{a}{2}(1-\cos\phi)} - \frac{\frac{a}{-1+a} + \frac{a}{2}(1+\cos\phi)}{\frac{1}{-1+a} + \frac{a}{2}(1+\cos\phi)} \right] + 2K_{\text{WT}}^{\text{Acos}\phi} - 2H_{\text{WT}}^{\text{Acos}\phi}$$

$$= \lim_{A \to 0} \frac{-2A\cos\phi}{-2H_{WT}} = 0/0^{\circ}$$
 for $0 \le \phi < \pi/2$
$$= 0/180^{\circ}$$
 for $\pi/2 < \phi \le \pi$

Similarly, for $\pi \leq \phi < 2\pi$,

$$\begin{array}{c} -2A\cos\varphi \\ A \rightarrow 0 \\ 3T_{FPO} \end{array} \underbrace{ \begin{bmatrix} \frac{-1 + \sqrt{a^2 + 1}}{-1 - a + \sqrt{a^2 + 1}} + \frac{a}{2}(1 - \cos\varphi) & \frac{-1 + \sqrt{a^2 + 1}}{-1 + a + a + 1} + \frac{a}{2}(1 - \cos\varphi) & \frac{-1 + \sqrt{a^2 + 1}}{-1 + a + \sqrt{a^2 + 1}} - \frac{a}{2}(1 + \cos\varphi) \\ \frac{a}{-1 + a + \sqrt{a^2 + 1}} + \frac{a}{2}(1 + \cos\varphi) & \frac{-1 + \sqrt{a^2 + 1}}{-1 + a + \sqrt{a^2 + 1}} + \frac{a}{2}(1 + \cos\varphi) \\ \end{array} \right) + \\ \end{array}$$

$$\frac{-2A\cos\phi}{A \to 0} = \lim_{A \to 0} \frac{-2A\cos\phi}{3T_{\text{FPO}} \left[\frac{\frac{a}{-1 + a} + \frac{a}{2}(1 - \cos\phi)}{\frac{1}{1 + a} + \frac{a}{2}(1 - \cos\phi)} - \frac{\frac{a}{1 + a} - \frac{a}{2}(1 + \cos\phi)}{\frac{1}{1 + a} + \frac{a}{2}(1 + \cos\phi)} \right] + 2K_{\text{WT}}A\cos\phi + 2H_{\text{WT}}$$

$$= \lim_{A \to 0} \frac{-2A\cos\phi}{2K_{\text{WT}}A\cos\phi + 2H_{\text{WT}}}$$

$$= 0/180^{\circ} + \tan^{-1} \frac{\cos\phi}{|\cos\phi|} = 0/180^{\circ} \qquad 3\pi/2 \le \phi < 2\pi$$

$$= 0/0^{\circ} \qquad \pi \le \phi < 3\pi/2 \quad \text{Q.E.D.}$$

4.4 The DDF of the Combined Nonlinearity For $N \ge 3$

A general relation can be derived for the combined nonlinearity for $N \geq 3$.

In general, the z-transform of the output of the combined nonlinearity may be written as

$$T_{q}(z) = \sum_{m=0}^{\infty} \sum_{k=0}^{N-1} T_{n}(kT) z^{-k-mN}$$

$$= \frac{\sum_{k=0}^{N-1} T_{n}(kT) z^{N-k}}{z^{N-1}}$$

$$= \frac{\sum_{k=0}^{N-1} T_{n}(kT) z^{N-k}}{z^{N-1}}$$
(4-52)

The discrete describing function then becomes

$$N(z) = \frac{T_{n}(z)}{\Theta_{i}(z)} = \frac{\sum_{k=0}^{N-1} T_{n}(kT) z^{N-k}}{(z^{N}-1) \sum_{k=0}^{\infty} (A\cos\frac{2\pi k}{N} + \phi) z^{-k}}$$
(4-53)

The last equation is simplified to

$$N(z) = \frac{\sum_{k=0}^{N-1} T_n(kT) z^{N-k-1}}{A \sum_{k=0}^{N-1} (\cos \frac{2\pi k}{N} + \phi) z^{N-k-1}}$$
 (N \geq 3) (4-54)

Or, alternately,

Or,
$$N(z) = \frac{\sum_{k=0}^{N-1} T_n(kT) z^{N-k-1}}{A(z-1) \sum_{k=2}^{N-2} (z-e^{j2\pi k/N}) ((z-\cos\frac{2\pi}{N})\cos\phi - \sin\frac{2\pi}{N}\sin\phi)}$$
(4-56)

For N = 3, $z = e^{j2\pi/3}$,

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$$N(z) = \frac{T_n(0)z^2 + T_n(T)z + T_n(2T)}{A(z-1)((z+0.5)\cos\phi - 0.866\sin\phi)}$$
(4-57)

Similar expressions can be obtained for N = 4, 5, . . . with z identified with $e^{j2\pi/N}$.

In general,

$$T_n(kT) = T_n^-(kT)$$
 $0 < \frac{2\pi k}{N} + \phi < \pi$ (4-58)
= $T_n^+(kT)$ $\pi < \frac{2\pi k}{N} + \phi < 2\pi$

4.5 Asymptotic Properties of -1/N(z) for $N \ge 3$ as A Approaches Infinity

The following properties of -1/N(z) are found for N \geq 3. The proofs of these properties can be obtained the same way as those for N = 2 by replacing ϕ by ϕ + $2\pi k/N$.

(a)
$$\lim_{A\to\infty} (-1/N(z)) = -\frac{1}{K_{WT}}$$
 for all ϕ (4-59)

(b)
$$\lim_{A\to\infty} |-1/N(z)| = \frac{1}{K_{WT}}$$
 for all ϕ (4-60)

(c)
$$\lim_{A\to\infty} \left(Arg(-1/N(z)) = 180^{\circ} \right)$$
 for all ϕ (4-61)

4.6 Asymptotic Properties of -1/N(z) for $N \ge 3$ as A Approaches Zero

The following properties of -1/N(z) for $N \ge 3$ are obtained as A

approaches zero.

(a) For $N \ge 3$, and for all ϕ ,

$$\lim_{A \to 0} |1/N(z)| = 0$$
 (4-62)

(b) For $N \ge 3$, (N = odd integers)

$$\lim_{A \to 0} Arg(-1/N(z)) = -\left(\frac{1+3N+2k}{2N}\right)\pi + \phi$$
 (4-63)

for $k\pi/N \le \phi < (k + 1)\pi/N$, k = 0, 1, 2, ..., (2N - 1).

(c) For $N \ge 4$, (N = even integers)

$$\lim_{A \to 0} Arg \left(-1/N(z)\right) = -\left(\frac{2 + 3N + 4k}{2N}\right)\pi + \phi \tag{4-64}$$

for $2k\pi/N \le \phi < 2(k + 1)\pi/N$.

In view of the properties of -1/N(z) listed above, the following theorems are generated.

Theorem 4-3.

For even integral N \geq 3, the magnitude and phase of -1/N(z) repeat for every ϕ = $2\pi/N$ radians.

Theorem 4-4.

For odd N (N \geq 3), the magnitude and phase of -1/N(z) repeat for every = π/N radians.

4.7 Discrete Describing Function Plots of the Combined Wire-Cable and Flex-Pivot Nonlinearity - The Critical Regions

The discrete describing function, N(z), for the combined wire-cable and flex-pivot nonlinearity is derived in the preceding sections. The plots of F(z) = -1/N(z) together with the plot of $G_{\rm eq}(z)$, which is the linear transfer function that N(z) sees, in the frequency domain allow the study of the condition of self-sustained oscillations of the digital IPS system.

Computer programs for the evaluation of the -1/N(z) for N=2 and $N\geq 3$ have been prepared. The listings of these programs are given in Tables 4-1 and 4-2, respectively.

For N = 2, the expression for F(z) = -1/N(z) is given in Eqs. (4-35), (4-36) and (4-37). Figure 4-3 shows the F(z) plot for N = 2 in the gain-phase coordinates with 0 < A < ∞ and all values of ϕ .

The following set of parameters are used for the nonlinear elements:

$$T_{FPO} = 0.00225$$
 $\gamma = 9.2444 \times 10^{4}$
 $K_{WT} = 100$
 $H_{WT} = 1$

In view of Theorem 4-1, Eq. (4-44), the most important parameter among those listed above is $K_{\rm WT}$, since when A approaches infinity the magnitude of F(z) approaches $1/K_{\rm WT}$. However, as shown in Fig. 4-3, the F(z) plot stays on the $-180^{\rm O}$ and $-360^{\rm O}$ axes for N = 2.

The discrete describing function N(z) for N \geq 3 is given by Eq. (4-54) or Eq. (4-56). Figure 4-4 shows the gain-phase plot of F(z) = -1/N(z) for N = 3. The curves for several values of ϕ between 0° and 60° are plotted to illustrate the effect of varying the phase of the input signal to the nonlinearity. It should be noted that for N = 3, Theorem 4-3 states that the values of F(z) repeat every 60 degrees starting from ϕ = 0°. As the magnitude of the input signal, A, approaches infinity, the magnitude of F(z) becomes $1/K_{WT}$ which is -40 db in this case, since K_{WT} is 100. On the other hand, as A approaches zero, the magnitude of F(z) becomes zero or - ∞ db, and the phase of F(z) is bounded by -300° and -240° for all values of ϕ . When the value of K_{WT} is varied, the curves of F(z) will shift up or down according to

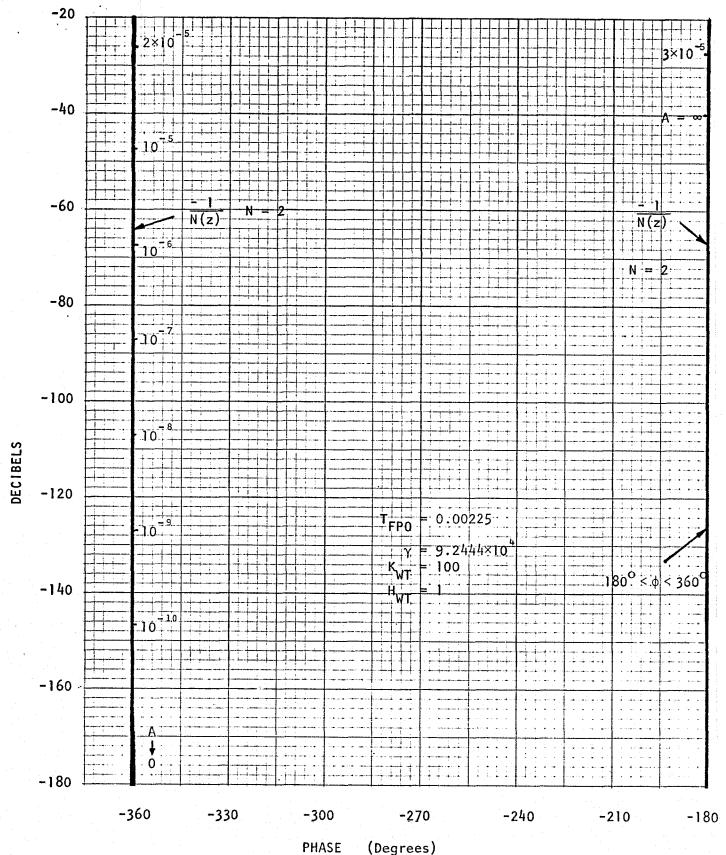
Table 4-1. Computer program for the computation of the discrete describing function of the combine flex-pivot and wire-cable nonlinearity of the IPS. N=2.

	of the IPS. $N = 2$.
··C	CALCULATION FOR -1/N(Z) FOR COMBINED NONLINEARITY, N=2
	COMPLEX GN1,GN2 REAL*8 P(20),RAD,TO,GAMMA,ASTART,A,AA,R,CSS,CSP,CSC
	REAL*8 AA2,R1,R2,R3,R4,TOF,TOM,TTM,TTF,T1,T2
	PI=3.14159D0
	RAD=180.DO/PI
	CSP=1.DO
	T0=0.00225D0
	KWT=100.DO
	HWT=1.00
	GAMMA=9.2444D4 REPRODUCI
	ASTART=1.0D-10 ORGINAL PAGE
	NP=5
	ND=15
	WRITE(5,100)
•	WRITE (5/101)
	DO 1 J=1,ND
	DO 1 I=1,NP
	CSS=1.DO-CSP
	CSC=1.DO+CSF
	A=ASTART*DFLOAT(I)*(10.DO**(J-1))
	AA=2.DO%GAMMA%A%TO
	R=(-1.DO/AA)+DSQRT((AA*AA+1.DO)/(AA*AA))
	AA2=AA/2.DO
	RI=1,D0/(R-1,D0)
	R2=R*R1
	R3=1.DO/(R+1.DO)
	R4=R×R3
	TOP=(R2+AA2*CSS)/(R1+AA2*CSS)
	TOM=(R4-AA2*CSS)/(R3+AA2*CSS)
	TTM=(R4-AA2*CSC)/(R3+AA2*CSC) TTP=(R2+AA2*CSC)/(R1+AA2*CSC)
	T1=3.DO*TO*(TOP-TTM)+2.DO*(HWT+KWT*AA*CSP)
	T2=3.DOXTOX(TON-TTP)+2.DOX(HWT+KWTXAAXCSF)
	GN1=-2.DO%AA%CSP/T1
	GN2=-2.DO%AA%CSP/T2
	G11=REAL(GN1)
	G12=AIMAG(GN1)
	G21=REAL(GN2)
	G22=AIMAG(GN2)
	GMAG1=CABS(GN1)
	GMAG2=CABS(GN2)
	GDB1=20.%ALOG10(GMAG1)
	GDB2=20.*AL0G10(GMAG2)
	GPH1=RAD*ATAN2(G12,G11)
	GPH2=RAD%ATAN2(G22,G21)
	IF(GPH2.GE.O.)GPH2=GPH2-340.
	IF(GFH1.GE.O.)GFH1=GFH1-360.
2.5	WRITE(5,102)A,GDB2,GMAG2,GFH2
1	CONTINUE
100	FORMAT(' DISCRETE DESCRIBING FUNCTION FOR IFS ')
101	FORMAT(//8X,/A/,11X,/GDB2/,10X,/GMAG/,9X,/FHASE//)
102	FORMAT (1F4E14.5)
	STOF
	END

Table 4-2. Computer program for the computation of the discrete describing function of the combined flex-pivot and wire-cable nonlinearity of the IPS. N GE. 3.

```
C
        DISCRETE DESCRIBING FUNCTION FOR COMBINED NONLINEARITY, N.GE.3
        REAL PHI:P(15):PI:RAD:GAMMA:ASTART:A:AAA:R:HWT:KWT:TO
        REAL PP:RR:AAR:R1:R2:R3:R4:AA2:TC:PHIK:PHID:PIK:TWN:TWP
        COMPLEX GV.TTSUM
        COMPLEX TSUMN, ZSUMN, ZSUM, TZ, TSUM, Z, THETA, GNN, GN
        PI=3.14159
        RAD=180./PI
        HWT=1.0
        KWT=100.
        59MMH=9.2444E4
        ASTART=1.E-8
        MD=9
        MP=1
        TU=0.00225
        RR=0.
                                            REPRODUCIELLARY OF THE
        MI = 50
                                            ORIGINAL PAGE IS STOP
        AM=FLOAT(NI)
        M1=MI-1
        M2=MI-2
        P(1)=1.0
        MPHI=3600
        PP=2.*PI/FLOAT(NI)
        THETA=CMPLX(RR,PP)
        Z=CEXP(THETA)
        ZSUMN=CMPLX(1.,0.0)
        TSUMN=CMPLX(D.0,0.0)
        WRITE(5,100)
        WRITE(5,102) AN
        WRITE(5,110) GAMMA
        WRITE(5,111) HWT
        WRITE(5,112) KWT
        WRITE(5,113) TO
        WRITE(5,101)
        DD 8 I=0,144,36
        PHI=(2.0*PI*FLOAT(I))/(FLOAT(NPHI))
        PHID=PHI*RED
        WRITE(5,103)PHID
        DJ 1 J=1.ND
        DJ 9 L=1 MP
        ZSUMN=CMPLX(1.0,0.0)
        TSUMN=CMPLX(0.0,0.0)
        A=ASTART+FLOAT(E)+(10.++(J-1))
        AA=2.0+GAMMA+A+TD
        995=88/2.0
```

```
R=(-1.0/AA)+SQRT((AA+AA+1.0)/(AA+AA))
        R1=1.0/(R-1.0)
        R2=R+R1
        R3=1.07(R+1.0)
        R4=R+R3
        DD 2 K=0.N1
        PIK=(2.0+PI+FLOAT(K))/(FLOAT(NI))
        PHIK=PHI+PIK
        IF(PHIK.GT.(2.0*PI))PHIK=PHIK-2.0*PI
        TWN=-HWT+KWT+A+COS(PHIK)
        TWP=HWT+KWT+A+COS(PHIK)
        IF(PHIK.LT.PI)50 TO 6
        TC=3.0+TO+(R2+AA2-AA2+COS(PHIK))/(R1+AA2-AA2+COS(PHIK))+TWP
        60 TD 7
        TC=3.0+TO+(R4-AA2+AA2+COS(PHIK))/(R3+AA2-AA2+COS(PHIK))+TWN
6
7
        TSUM=TC*Z**(N1-K)
        TSUMM=TSUMM+TSUM
\Xi
        CONTINUE
        IF(NI.LE.3)60 TO 10
        DO 3 M=2.N2
        Z$UM=Z-Z++M
        ZSUMM=ZSUMM*ZSUM
3
        CONTINUE
10
        CONTINUE
        TZ=(Z-COS(PP))+COS(PHI)-SIN(PP)+SIN(PHI)
        GM=TSUMM/(A+TZ+(Z-1.)+ZSUMM)
        6MM=-1./6M
        GV=GNN
        G1=REAL(GV)
        S2=AIMAS(GV)
        GMAG=CABS(GV)
        GDB=20.+ALO610(6MAG)
        GPHASE=RAD+ATAN2(62,61)
        IF(GPHASE.GE.0.)GPHASE=GPHASE-360.
        WRITE(5,104)A,GPHASE,GDB,GMAG
105
        FORMAT(1PE24.15)
9
        CONTINUE
1
        CONTINUE
8
        CONTINUE
100
        FORMAT(7X)/DESCRIBING FUNCTION OF COMBINED MONLINEARITY()
101
        FDRMAT(/,7X,'A',10X,'PHASE(,10X,'DB',10X,'MAGNITUDE()
102
        FORMAT(5X, 1N=1, F4.1)
110
        FORMAT(/,5%,/GAMMA=/,19E12.4)
111
        FORMAT(5X,/HWT=/,F8.2)
112
        FORMAT(5X,/KWT=/,F8.2)
113
        FORMAT(5X, 'TO=', 1PE12.2)
103
        FDRMAT(/,5X,'PHI=/,F5.1)
104
        FORMAT(195E14.5)
        STUP
        END
```



 $L_{t_{1}}$

Figure 4-3. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N = 2.

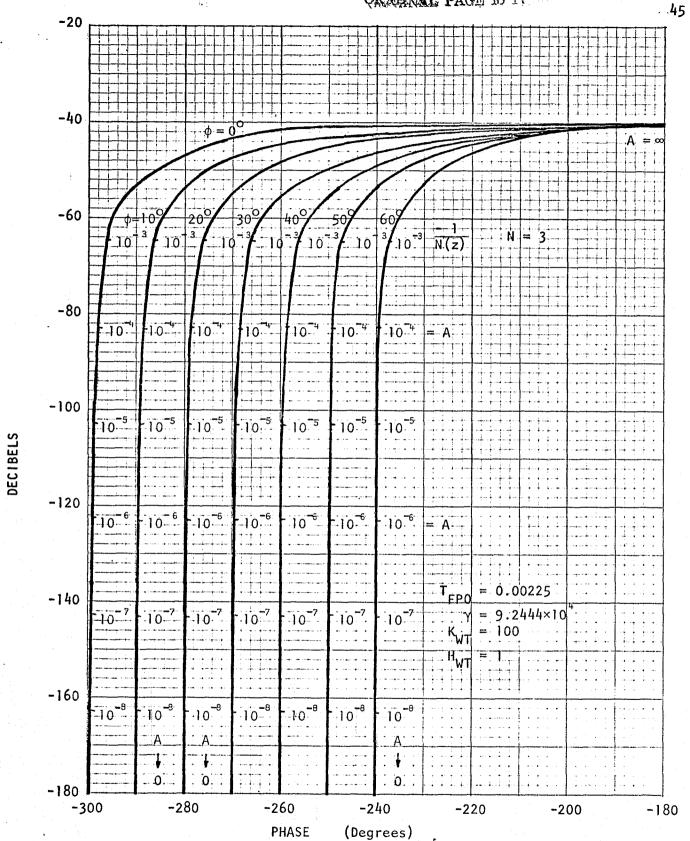


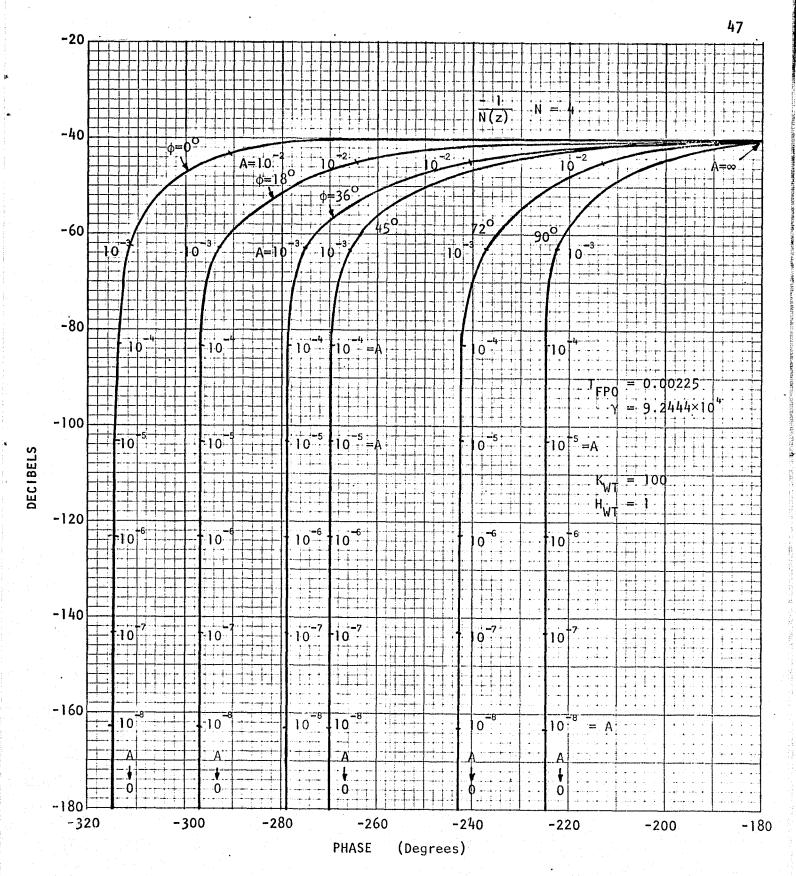
Figure 4-4. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N=3.

 $1/K_{WT}$ in db; when the values of T_{FPO} , γ , and H_{WT} are varied, the shape of the curved portions of the plots in Fig. 4-4 will be changed. However, in general, the impact of the variation of K_{WT} will be the greatest.

The F(z) plots for N = 4, 5,6,8,10, 20, and 50 are shown in Figs. 4-5 through 4-11, respectively. For N = 4, the F(z) plot extends from -315° to -225° as ϕ varies. For N = 5, the span of the plot is -288° to -252° .

For stability analysis, it is sufficient to consider only the bounds of the F(z) plot for a fixed N. Self-sustained oscillations characterized by the period $T_c = NT$, where T is the sampling period, may occur if $G_{eq}(z)$ intersects with any part of the F(z) plot. The region bounded by all the F(z) curves for a given N is defined as the <u>critical region</u>. The critical regions for the combined nonlinearity for N > 2 are the regions that are bounded by the F(z) curves for $\phi = 0^O$ and $\phi = 2\pi/N$ for N = even and $\phi = \pi/N$ for N = odd.

As N approaches infinity, the discrete describing function N(z) approaches the describing function N of the continuous-data nonlinearity, as shown in Fig. 4-11. It is observed that as N increases the width of the critical region becomes narrower. As N approaches infinity, the critical region of F(z) = -1/N(z) approaches the -1/N plot shown in Fig. 2-1.



.Figure 4-5. Discrete Describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N=4.

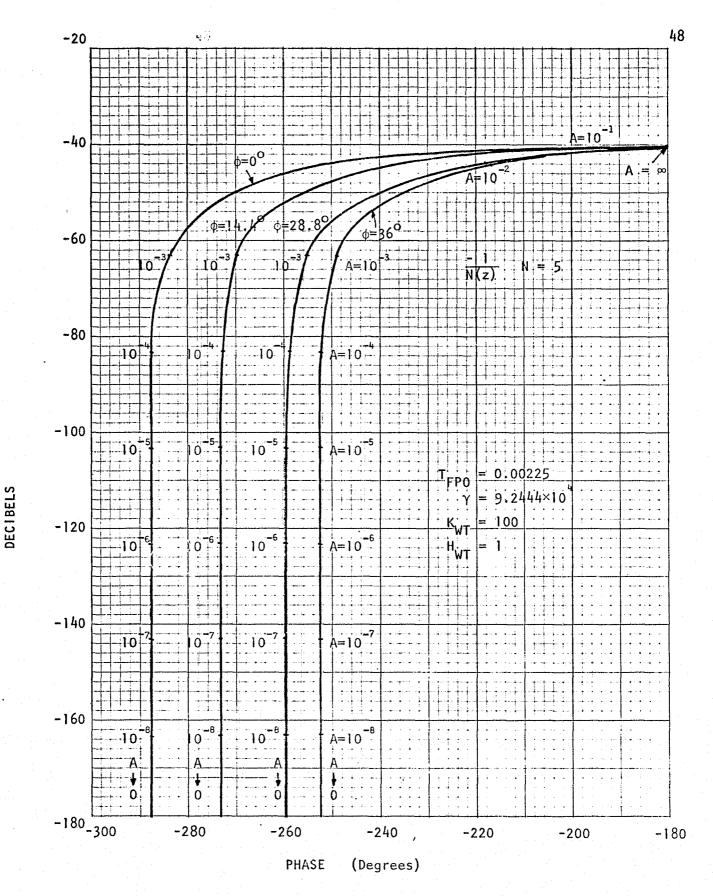


Figure 4-6. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N=5.

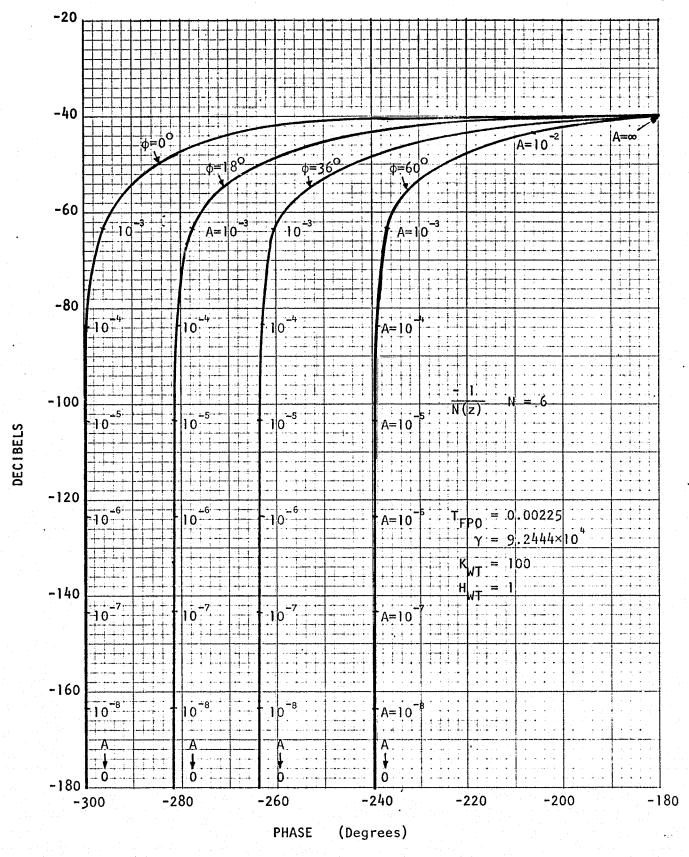


Figure 4-7. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N=6.

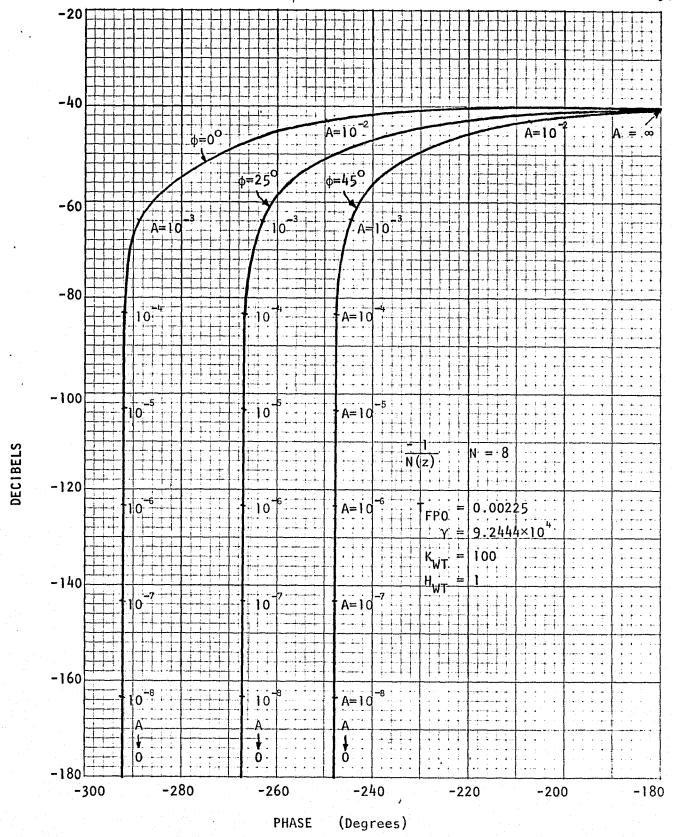


Figure 4-8. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N=8.

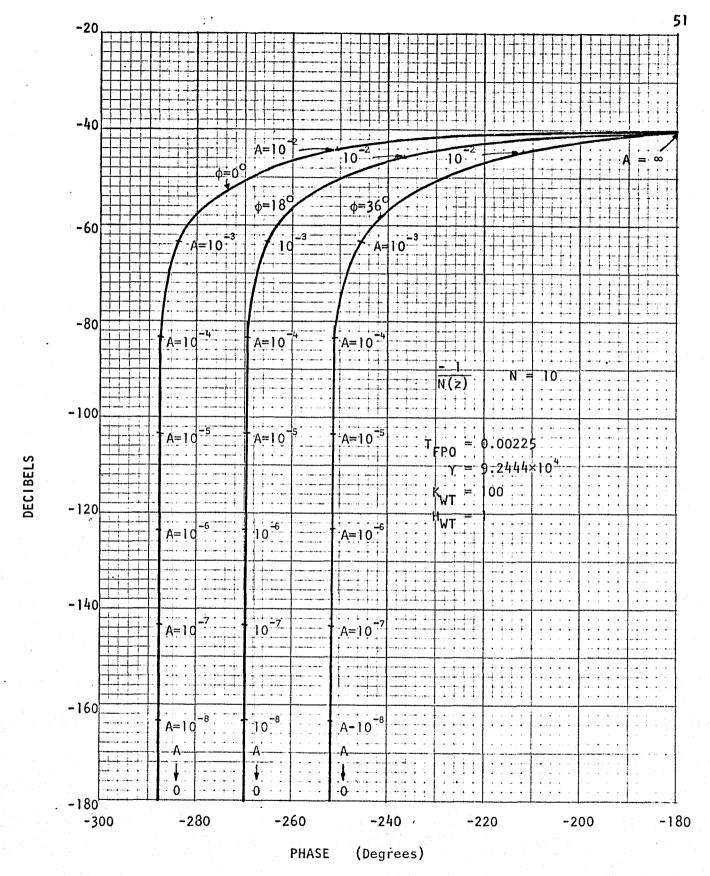


Figure 4-9. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N = 10.

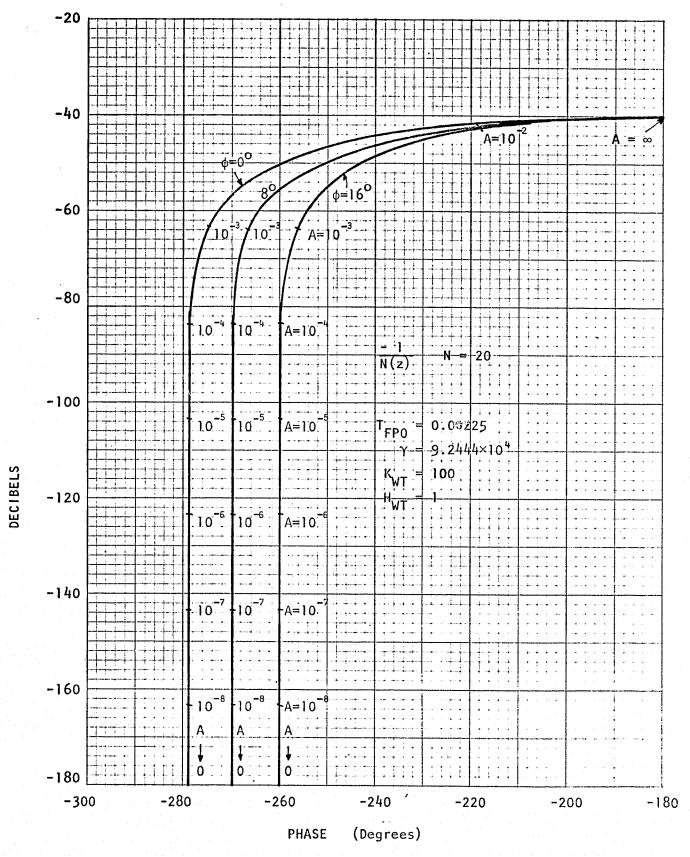


Figure 4-10. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N = 20.

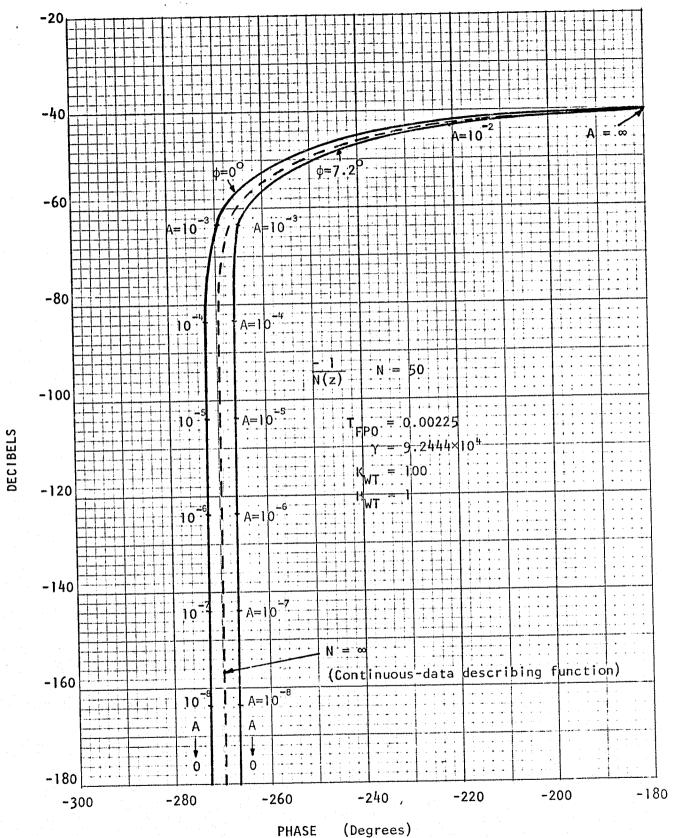


Figure 4-11. Discrete describing function plots of combined flex-pivot and wire-cable nonlinearity for the IPS. N=50.

REFERENCES

Final Report, Research Study on IPS Digital Controller Design, V-76,
 Systems Research Laboratory, September 1, 1976.